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## **DESIGN CALCULATION**

*In Accordance with ASME Section VIII Division 1*

ASME Code Version : 2017

Analysis Performed by : SPLM Licensed User

Job File : D:\2-PAYAM\1-PROJECTS\MICROTEC PROJECT\13-MICROT

Date of Analysis : Feb 20,2026 4:14pm

PV Elite 2019 SP1, March 2019

Note:

PV Elite performs all calculations internally in Imperial Units to remain compliant with the ASME Code and any built in assumptions in the ASME Code formulas. The finalized results are reflected to show the user's set of selected units.

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FileName : Circular Sections -----

Warnings and Errors: Step: 0 4:14pm Feb 20, 2026

Class From To : Basic Element Checks.

=====

Class From To: Check of Additional Element Data

=====

There were no geometry errors or warnings.

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FileName : Circular Sections

Input Echo:

Step: 1 4:14pm Feb 20,2026

**PV Elite Vessel Analysis Program: Input Data**

Design Internal Pressure (for Hydrotest)	0.5	bars
Design Internal Temperature	170.0	°C
Type of Hydrotest	not Specified	
Hydrotest Position	Horizontal	
Projection of Nozzle from Vessel Top	0	mm.
Projection of Nozzle from Vessel Bottom	0	mm.
Minimum Design Metal Temperature	-28.9	°C
Type of Construction	Welded	
Special Service	None	
Degree of Radiography	RT-1	
Use Higher Longitudinal Stresses (Flag)	Y	
Select t for Internal Pressure (Flag)	N	
Select t for External Pressure (Flag)	N	
Select t for Axial Stress (Flag)	N	
Select Location for Stiff. Rings (Flag)	N	
Consider Vortex Shedding	N	
Perform a Corroded Hydrotest	N	

Load Case 1	NP+EW+WI+FW+BW
Load Case 2	NP+EW+EE+FS+BS
Load Case 3	NP+OW+WI+FW+BW
Load Case 4	NP+OW+EQ+FS+BS
Load Case 5	NP+HW+HI
Load Case 6	NP+HW+HE
Load Case 7	IP+OW+WI+FW+BW
Load Case 8	IP+OW+EQ+FS+BS
Load Case 9	EP+OW+WI+FW+BW
Load Case 10	EP+OW+EQ+FS+BS
Load Case 11	HP+HW+HI
Load Case 12	HP+HW+HE
Load Case 13	IP+WE+EW
Load Case 14	IP+WF+CW
Load Case 15	IP+VO+OW
Load Case 16	IP+VE+EW
Load Case 17	NP+VO+OW
Load Case 18	FS+BS+IP+OW
Load Case 19	FS+BS+EP+OW

Wind Design Code	ASCE-7 93
Basic Wind Speed	[V] 112.65 Km/hr
Surface Roughness Category	C: Open Terrain
Importance Factor	1.0
Type of Surface	Moderately Smooth
Base Elevation	0 mm.
Percent Wind for Hydrotest	33.0
Using User defined Wind Press. Vs Elev.	N
Damping Factor (Beta) for Wind (Ope)	0.0100
Damping Factor (Beta) for Wind (Empty)	0.0000
Damping Factor (Beta) for Wind (Filled)	0.0000

Seismic Design Code	UBC 94
UBC Seismic Zone (1=1,2=2a,3=2b,4=3,5=4)	0.000
UBC Importance Factor	1.000
UBC Soil Type	S1
UBC Horizontal Force Factor	3.000
UBC Percent Seismic for Hydrotest	0.000

Design Pressure + Static Head	Y
Consider MAP New and Cold in Noz. Design	N
Consider External Loads for Nozzle Des.	Y
Use ASME VIII-1 Appendix 1-9	N

FileName : Circular Sections

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Material Database Year Current w/Addenda or Code Year

**Configuration Directives:**

Do not use Nozzle MDMT Interpretation VIII-1 01-37	No
Use Table G instead of exact equation for "A"	Yes
Shell Head Joints are Tapered	Yes
Compute "K" in corroded condition	Yes
Use Code Case 2286	No
Use the MAWP to compute the MDMT	Yes
For thickness ratios $\leq 0.35$ , MDMT will be -155F (-104C)	Yes
For PWHT & P1 Materials the MDMT can be $< -55F (-48C)$	No
Using Metric Material Databases, ASME II D	No
Calculate B31.3 type stress for Nozzles with Loads	Yes
Reduce the MDMT due to lower membrane stress	Yes
Consider Longitudinal Stress in MDMT calcs. (Div. 1)	No

**Complete Listing of Vessel Elements and Details:**

Element From Node	10	
Element To Node	20	
Element Type	Cylinder	
Description		
Distance "FROM" to "TO"	8052	mm.
Inside Diameter	1300	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Allowable Stress, Ambient	137.9	N./mm <sup>2</sup>
Allowable Stress, Operating	104.16	N./mm <sup>2</sup>
Allowable Stress, Hydrotest	179.27	N./mm <sup>2</sup>
Material Density	0.008027	kg./cm <sup>3</sup>
P Number Thickness	0	mm.
Yield Stress, Operating	156.11	N./mm <sup>2</sup>
External Pressure Chart Name	HA-2	
UNS Number	S31600	
Product Form	Plate	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Weld is pre-Heated	No	

Element From Node	10	
Detail Type	Ring	
Detail ID	Ring:[1 of 1]	
Dist. from "FROM" Node / Offset dist	3958	mm.
Inside Diameter of Ring	1306	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1400	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	10	
Detail Type	Ring	
Detail ID	Ring:[2 of 2]	
Dist. from "FROM" Node / Offset dist	5946	mm.
Inside Diameter of Ring	1306	mm.
Thickness of Ring	10	mm.

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Outside Diameter of Ring	1400	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	10	
Detail Type	Ring	
Detail ID	Ring:[3 of 3]	
Dist. from "FROM" Node / Offset dist	7939	mm.
Inside Diameter of Ring	1306	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1400	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	10	
Detail Type	Ring	
Detail ID	Ring:[4 of 4]	
Dist. from "FROM" Node / Offset dist	7952	mm.
Inside Diameter of Ring	1306	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1400	mm.
Material Name	SA-240 316	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	20	
Element To Node	30	
Element Type	Conical	
Description		
Distance "FROM" to "TO"	2000	mm.
Inside Diameter	1300	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Cone Diameter at "To" End	1730	mm.
Design Length of Cone	2000	mm.
Half Apex Angle of Cone	6.1357331	degrees
Toriconical (Y/N)	N	
Weld is pre-Heated	No	

Element From Node	20	
Detail Type	Ring	
Detail ID	Ring 16	
Dist. from "FROM" Node / Offset dist	0	mm.
Inside Diameter of Ring	1306	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1500	mm.
Material Name	SA-240 304	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

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FileName : Circular Sections

Input Echo: Step: 1 4:14pm Feb 20,2026

Element From Node	30
Element To Node	40
Element Type	Cylinder
Description	
Distance "FROM" to "TO"	800 mm.
Inside Diameter	1730 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Weld is pre-Heated	No

Element From Node	30
Detail Type	Ring
Detail ID	Ring5
Dist. from "FROM" Node / Offset dist	0 mm.
Inside Diameter of Ring	1736 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1830 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	40
Element To Node	50
Element Type	Conical
Description	
Distance "FROM" to "TO"	1200 mm.
Inside Diameter	1730 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Cone Diameter at "To" End	1400 mm.
Design Length of Cone	1200 mm.
Half Apex Angle of Cone	7.8290772 degrees
Toriconical (Y/N)	N
Weld is pre-Heated	No

Element From Node	40
Detail Type	Ring
Detail ID	Ring:17
Dist. from "FROM" Node / Offset dist	1199 mm.
Inside Diameter of Ring	1406.3 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.



FileName : Circular Sections

Input Echo:

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Using Custom Stiffener Section	No
Element From Node	40
Detail Type	Ring
Detail ID	Ring 21
Dist. from "FROM" Node / Offset dist	0 mm.
Inside Diameter of Ring	1736 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1836 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	50
Element To Node	60
Element Type	Cylinder
Description	
Distance "FROM" to "TO"	4500 mm.
Inside Diameter	1400 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Weld is pre-Heated	No

Element From Node	50
Detail Type	Ring
Detail ID	Ring 6
Dist. from "FROM" Node / Offset dist	90 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	50
Detail Type	Ring
Detail ID	Ring 7
Dist. from "FROM" Node / Offset dist	103 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	50
Detail Type	Ring
Detail ID	Ring 8
Dist. from "FROM" Node / Offset dist	1513 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.

FileName : Circular Sections

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Using Custom Stiffener Section	No
Element From Node	50
Detail Type	Ring
Detail ID	Ring 9
Dist. from "FROM" Node / Offset dist	2913 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	60
Element To Node	70
Element Type	Conical
Description	
Distance "FROM" to "TO"	2000 mm.
Inside Diameter	1400 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Cone Diameter at "To" End	1730 mm.
Design Length of Cone	2000 mm.
Half Apex Angle of Cone	4.7162204 degrees
Toriconical (Y/N)	N
Weld is pre-Heated	No

Element From Node	60
Detail Type	Ring
Detail ID	Ring 18
Dist. from "FROM" Node / Offset dist	0 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	60
Detail Type	Ring
Detail ID	Ring 23
Dist. from "FROM" Node / Offset dist	1999 mm.
Inside Diameter of Ring	1735.8 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1829.8 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	70
Element To Node	80
Element Type	Cylinder

FileName : Circular Sections

Input Echo:

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Description	
Distance "FROM" to "TO"	800 mm.
Inside Diameter	1730 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Weld is pre-Heated	No

Element From Node	70
Detail Type	Ring
Detail ID	Ring 10
Dist. from "FROM" Node / Offset dist	737 mm.
Inside Diameter of Ring	1736 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1830 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	70
Detail Type	Ring
Detail ID	Ring 11
Dist. from "FROM" Node / Offset dist	750 mm.
Inside Diameter of Ring	1736 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1830 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

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Element From Node	80
Element To Node	90
Element Type	Conical
Description	
Distance "FROM" to "TO"	1200 mm.
Inside Diameter	1730 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Cone Diameter at "To" End	1400 mm.
Design Length of Cone	1200 mm.
Half Apex Angle of Cone	7.8290772 degrees
Toriconical (Y/N)	N
Weld is pre-Heated	No

Element From Node	80
-------------------	----

FileName : Circular Sections

Input Echo:

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Detail Type	Ring
Detail ID	Ring 24
Dist. from "FROM" Node / Offset dist	0 mm.
Inside Diameter of Ring	1736 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1829.7 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	80
Detail Type	Ring
Detail ID	Ring 26
Dist. from "FROM" Node / Offset dist	1199 mm.
Inside Diameter of Ring	1406.3 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1499.9 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

---

Element From Node	90
Element To Node	100
Element Type	Cylinder
Description	
Distance "FROM" to "TO"	3500 mm.
Inside Diameter	1400 mm.
Element Thickness	3 mm.
Internal Corrosion Allowance	0 mm.
Nominal Thickness	3 mm.
External Corrosion Allowance	0 mm.
Design Internal Pressure	0.5 bars
Design Temperature Internal Pressure	170 °C
Design External Pressure	0.1 bars
Design Temperature External Pressure	50 °C
Effective Diameter Multiplier	1.2
Material Name	SA-240 316
Efficiency, Longitudinal Seam	1.0
Efficiency, Circumferential Seam	1.0
Weld is pre-Heated	No

Element From Node	90
Detail Type	Ring
Detail ID	Ring 12
Dist. from "FROM" Node / Offset dist	375 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

Element From Node	90
Detail Type	Ring
Detail ID	Ring 13]
Dist. from "FROM" Node / Offset dist	1745 mm.
Inside Diameter of Ring	1406 mm.
Thickness of Ring	10 mm.
Outside Diameter of Ring	1500 mm.
Material Name	SA-240 304
Height of Section Ring	0 mm.
Using Custom Stiffener Section	No

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Input Echo: Step: 1 4:14pm Feb 20,2026

Element From Node	100	
Element To Node	110	
Element Type	Conical	
Description		
Distance "FROM" to "TO"	2000	mm.
Inside Diameter	1400	mm.
Element Thickness	3	mm.
Internal Corrosion Allowance	0	mm.
Nominal Thickness	3	mm.
External Corrosion Allowance	0	mm.
Design Internal Pressure	0.5	bars
Design Temperature Internal Pressure	170	°C
Design External Pressure	0.1	bars
Design Temperature External Pressure	50	°C
Effective Diameter Multiplier	1.2	
Material Name	SA-240 316	
Efficiency, Longitudinal Seam	1.0	
Efficiency, Circumferential Seam	1.0	
Cone Diameter at "To" End	950	mm.
Design Length of Cone	2000	mm.
Half Apex Angle of Cone	6.4187856	degrees
Toriconical (Y/N)	N	
Weld is pre-Heated	No	

Element From Node	100	
Detail Type	Ring	
Detail ID	Ring 22	
Dist. from "FROM" Node / Offset dist	1999	mm.
Inside Diameter of Ring	956.22	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1057.1	mm.
Material Name	SA-240 304	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

Element From Node	100	
Detail Type	Ring	
Detail ID	Ring 28	
Dist. from "FROM" Node / Offset dist	0	mm.
Inside Diameter of Ring	1406	mm.
Thickness of Ring	10	mm.
Outside Diameter of Ring	1506.8	mm.
Material Name	SA-240 304	
Height of Section Ring	0	mm.
Using Custom Stiffener Section	No	

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FileName : Circular Sections

XY Coordinate Calculations: Step: 2 4:14pm Feb 20,2026

**XY Coordinate Calculations:**

From	To	X (Horiz.) mm.	Y (Vert.) mm.	DX (Horiz.) mm.	DY (Vert.) mm.
10	20	...	8052	...	8052
20	30	...	10052	...	2000
30	40	...	10852	...	800
40	50	...	12052	...	1200
50	60	...	16552	...	4500
60	70	...	18552	...	2000
70	80	...	19352	...	800
80	90	...	20552	...	1200
90	100	...	24052	...	3500
100	110	...	26052	...	2000

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Internal Pressure Calculations: Step: 13 4:14pm Feb 20,2026

**Element Thickness, Pressure, Diameter and Allowable Stress :**

From	To	Int. Press + Liq. Hd bars	Nominal Thickness mm.	Total Corr Allowance mm.	Element Diameter mm.	Allowable Stress (SE) N./mm <sup>2</sup>
10	20	0.5	3	...	1300	104.16
20	30	0.5	3	...	1730	104.16
30	40	0.5	3	...	1730	104.16
40	50	0.5	3	...	1730	104.16
50	60	0.5	3	...	1400	104.16
60	70	0.5	3	...	1730	104.16
70	80	0.5	3	...	1730	104.16
80	90	0.5	3	...	1730	104.16
90	100	0.5	3	...	1400	104.16
100	110	0.5	3	...	1400	104.16

**Element Required Thickness and MAWP :**

From	To	Design Pressure bars	M.A.W.P. Corroded bars	M.A.P. New & Cold bars	Minimum Thickness mm.	Required Thickness mm.
10	20	0.5	4.79375	6.34667	3	1.5
20	30	0.5	3.58411	4.74517	3	1.5
30	40	0.5	3.60472	4.77245	3	1.5
40	50	0.5	3.57118	4.72806	3	1.5
50	60	0.5	4.45222	5.8945	3	1.5
60	70	0.5	3.59254	4.75633	3	1.5
70	80	0.5	3.60472	4.77245	3	1.5
80	90	0.5	3.57118	4.72806	3	1.5
90	100	0.5	4.45222	5.8945	3	1.5
100	110	0.5	4.42438	5.85764	3	1.5
Minimum			3.571	4.728		

MAWP: 3.571 bars, limited by: Cone.

**Internal Pressure Calculation Results :****ASME Code, Section VIII Division 1, 2017****Cylindrical Shell From 10 To 20 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)} \\
 &= (0.5 \cdot 650.0) / (104.16 \cdot 1.0 - 0.6 \cdot 0.5) \\
 &= 0.3121 + 0.0000 = 0.3121 \text{ mm.}
 \end{aligned}$$

*Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.*

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (104.16 \cdot 1.0 \cdot 3.0) / (650.0 + 0.6 \cdot 3.0) \\
 &= 4.794 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (650.0 + 0.6 \cdot 3.0)
 \end{aligned}$$

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Internal Pressure Calculations: Step: 13 4:14pm Feb 20, 2026

$$= 6.347 \text{ bars}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned} &= (P * (R + 0.6 * t)) / (E * t) \\ &= (0.5 * (650.0 + 0.6 * 3.0)) / (1.0 * 3.0) \\ &= 10.864 \text{ N./mm}^2 \end{aligned}$$

% Elongation per Table UG-79-1  $(50 * t_{nom} / R_f) * (1 - R_f / R_o)$  0.230 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

### **Conical Section From 20 To 30 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned} &= (P * D) / (2 * \cos(a) * (S * E - 0.6 * P)) \text{ per Appendix 1-4 (e)} \\ &= (0.5 * 1730.0) / (2 * 0.9943 * (104.16 * 1.0 - 0.6 * 0.5)) \\ &= 0.4178 + 0.0000 = 0.4178 \text{ mm.} \end{aligned}$$

*Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.*

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned} &= (2 * S * E * t * \cos(a)) / (D + 1.2 * t * \cos(a)) \text{ per App 1-4 (e)} \\ &= (2 * 104.16 * 1.0 * 3.0 * 0.994) / (1730.0 + 1.2 * 3.0 * 0.994) \\ &= 3.584 \text{ bars} \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned} &= (2 * S * E * t * \cos(a)) / (D + 1.2 * t * \cos(a)) \text{ per App 1-4 (e)} \\ &= (2 * 137.9 * 1.0 * 3.0 * 0.9943) / (1730.0 + 1.2 * 3.0 * 0.9943) \\ &= 4.745 \text{ bars} \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned} &= (P * (D + 1.2 * t * \cos(a))) / (2 * E * t * \cos(a)) \\ &= (0.5 * (1730.0 + 1.2 * 3.0 * 0.9943)) / (2 * 1.0 * 3.0 * 0.9943) \\ &= 14.531 \text{ N./mm}^2 \end{aligned}$$

% Elongation per Table UG-79-1  $(50 * t_{nom} / R_f) * (1 - R_f / R_o)$  0.230 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

*Note: The Pressure at the Large Diameter is used in the TR calculation.*

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

### **Cylindrical Shell From 30 To 40 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned} &= (P * R) / (S * E - 0.6 * P) \text{ per UG-27 (c) (1)} \\ &= (0.5 * 865.0) / (104.16 * 1.0 - 0.6 * 0.5) \\ &= 0.4154 + 0.0000 = 0.4154 \text{ mm.} \end{aligned}$$

*Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.*

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned} &= (S * E * t) / (R + 0.6 * t) \text{ per UG-27 (c) (1)} \\ &= (104.16 * 1.0 * 3.0) / (865.0 + 0.6 * 3.0) \\ &= 3.605 \text{ bars} \end{aligned}$$



FileName : Circular Sections

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Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (865.0 + 0.6 \cdot 3.0) \\
 &= 4.772 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t) \\
 &= (0.5 \cdot (865.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0) \\
 &= 14.448 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1  $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$  0.173 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

**Conical Section From 40 To 50 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot D) / (2 \cdot \cos(a) \cdot (S \cdot E - 0.6 \cdot P)) \text{ per Appendix 1-4 (e)} \\
 &= (0.5 \cdot 1730.0) / (2 \cdot 0.9907 \cdot (104.16 \cdot 1.0 - 0.6 \cdot 0.5)) \\
 &= 0.4193 + 0.0000 = 0.4193 \text{ mm.}
 \end{aligned}$$

*Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.*

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 \cdot 104.16 \cdot 1.0 \cdot 3.0 \cdot 0.991) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.991) \\
 &= 3.571 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 \cdot 137.9 \cdot 1.0 \cdot 3.0 \cdot 0.9907) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.9907) \\
 &= 4.728 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (D + 1.2 \cdot t \cdot \cos(a))) / (2 \cdot E \cdot t \cdot \cos(a)) \\
 &= (0.5 \cdot (1730.0 + 1.2 \cdot 3.0 \cdot 0.9907)) / (2 \cdot 1.0 \cdot 3.0 \cdot 0.9907) \\
 &= 14.583 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1  $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$  0.214 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

**Cylindrical Shell From 50 To 60 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)} \\
 &= (0.5 \cdot 700.0) / (104.16 \cdot 1.0 - 0.6 \cdot 0.5) \\
 &= 0.3361 + 0.0000 = 0.3361 \text{ mm.}
 \end{aligned}$$

*Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.*

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (104.16 \cdot 1.0 \cdot 3.0) / (700.0 + 0.6 \cdot 3.0)
 \end{aligned}$$

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$$= 4.452 \text{ bars}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned} &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\ &= (137.9 \cdot 1.0 \cdot 3.0) / (700.0 + 0.6 \cdot 3.0) \\ &= 5.894 \text{ bars} \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned} &= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t) \\ &= (0.5 \cdot (700.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0) \\ &= 11.697 \text{ N./mm}^2 \end{aligned}$$

% Elongation per Table UG-79-1  $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$  0.214 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

### **Conical Section From 60 To 70 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned} &= (P \cdot D) / (2 \cdot \cos(a) \cdot (S \cdot E - 0.6 \cdot P)) \text{ per Appendix 1-4 (e)} \\ &= (0.5 \cdot 1730.0) / (2 \cdot 0.9966 \cdot (104.16 \cdot 1.0 - 0.6 \cdot 0.5)) \\ &= 0.4168 + 0.0000 = 0.4168 \text{ mm.} \end{aligned}$$

*Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.*

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned} &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\ &= (2 \cdot 104.16 \cdot 1.0 \cdot 3.0 \cdot 0.997) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.997) \\ &= 3.593 \text{ bars} \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned} &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\ &= (2 \cdot 137.9 \cdot 1.0 \cdot 3.0 \cdot 0.9966) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.9966) \\ &= 4.756 \text{ bars} \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned} &= (P \cdot (D + 1.2 \cdot t \cdot \cos(a))) / (2 \cdot E \cdot t \cdot \cos(a)) \\ &= (0.5 \cdot (1730.0 + 1.2 \cdot 3.0 \cdot 0.9966)) / (2 \cdot 1.0 \cdot 3.0 \cdot 0.9966) \\ &= 14.496 \text{ N./mm}^2 \end{aligned}$$

% Elongation per Table UG-79-1  $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$  0.214 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

*Note: The Pressure at the Large Diameter is used in the TR calculation.*

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

### **Cylindrical Shell From 70 To 80 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned} &= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)} \\ &= (0.5 \cdot 865.0) / (104.16 \cdot 1.0 - 0.6 \cdot 0.5) \\ &= 0.4154 + 0.0000 = 0.4154 \text{ mm.} \end{aligned}$$

*Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.*

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Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)}$$

$$= (104.16 \cdot 1.0 \cdot 3.0) / (865.0 + 0.6 \cdot 3.0)$$

$$= 3.605 \text{ bars}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)}$$

$$= (137.9 \cdot 1.0 \cdot 3.0) / (865.0 + 0.6 \cdot 3.0)$$

$$= 4.772 \text{ bars}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t)$$

$$= (0.5 \cdot (865.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0)$$

$$= 14.448 \text{ N./mm}^2$$

% Elongation per Table UG-79-1  $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$  0.173 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

### **Conical Section From 80 To 90 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$= (P \cdot D) / (2 \cdot \cos(a) \cdot (S \cdot E - 0.6 \cdot P)) \text{ per Appendix 1-4 (e)}$$

$$= (0.5 \cdot 1730.0) / (2 \cdot 0.9907 \cdot (104.16 \cdot 1.0 - 0.6 \cdot 0.5))$$

$$= 0.4193 + 0.0000 = 0.4193 \text{ mm.}$$

*Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.*

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)}$$

$$= (2 \cdot 104.16 \cdot 1.0 \cdot 3.0 \cdot 0.991) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.991)$$

$$= 3.571 \text{ bars}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)}$$

$$= (2 \cdot 137.9 \cdot 1.0 \cdot 3.0 \cdot 0.9907) / (1730.0 + 1.2 \cdot 3.0 \cdot 0.9907)$$

$$= 4.728 \text{ bars}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$= (P \cdot (D + 1.2 \cdot t \cdot \cos(a))) / (2 \cdot E \cdot t \cdot \cos(a))$$

$$= (0.5 \cdot (1730.0 + 1.2 \cdot 3.0 \cdot 0.9907)) / (2 \cdot 1.0 \cdot 3.0 \cdot 0.9907)$$

$$= 14.583 \text{ N./mm}^2$$

% Elongation per Table UG-79-1  $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$  0.214 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

### **Cylindrical Shell From 90 To 100 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$= (P \cdot R) / (S \cdot E - 0.6 \cdot P) \text{ per UG-27 (c) (1)}$$

$$= (0.5 \cdot 700.0) / (104.16 \cdot 1.0 - 0.6 \cdot 0.5)$$

$$= 0.3361 + 0.0000 = 0.3361 \text{ mm.}$$

*Note: The thickness required was less than the Code Minimum, therefore*

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*the Code Minimum value of 1.5000 mm. per UG-16 will be used.*

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (104.16 \cdot 1.0 \cdot 3.0) / (700.0 + 0.6 \cdot 3.0) \\
 &= 4.452 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (S \cdot E \cdot t) / (R + 0.6 \cdot t) \text{ per UG-27 (c) (1)} \\
 &= (137.9 \cdot 1.0 \cdot 3.0) / (700.0 + 0.6 \cdot 3.0) \\
 &= 5.894 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (R + 0.6 \cdot t)) / (E \cdot t) \\
 &= (0.5 \cdot (700.0 + 0.6 \cdot 3.0)) / (1.0 \cdot 3.0) \\
 &= 11.697 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1  $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$  0.214 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

**Conical Section From 100 To 110 SA-240 316 at 170 °C**

Material UNS Number: S31600

Required Thickness due to Internal Pressure [tr]:

$$\begin{aligned}
 &= (P \cdot D) / (2 \cdot \cos(a) \cdot (S \cdot E - 0.6 \cdot P)) \text{ per Appendix 1-4 (e)} \\
 &= (0.5 \cdot 1400.0) / (2 \cdot 0.9937 \cdot (104.16 \cdot 1.0 - 0.6 \cdot 0.5)) \\
 &= 0.3383 + 0.0000 = 0.3383 \text{ mm.}
 \end{aligned}$$

*Note: The thickness required was less than the Code Minimum, therefore the Code Minimum value of 1.5000 mm. per UG-16 will be used.*

Max. Allowable Working Pressure at given Thickness, corroded [MAWP]:

$$\begin{aligned}
 &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 \cdot 104.16 \cdot 1.0 \cdot 3.0 \cdot 0.994) / (1400.0 + 1.2 \cdot 3.0 \cdot 0.994) \\
 &= 4.424 \text{ bars}
 \end{aligned}$$

Maximum Allowable Pressure, New and Cold [MAPNC]:

$$\begin{aligned}
 &= (2 \cdot S \cdot E \cdot t \cdot \cos(a)) / (D + 1.2 \cdot t \cdot \cos(a)) \text{ per App 1-4 (e)} \\
 &= (2 \cdot 137.9 \cdot 1.0 \cdot 3.0 \cdot 0.9937) / (1400.0 + 1.2 \cdot 3.0 \cdot 0.9937) \\
 &= 5.858 \text{ bars}
 \end{aligned}$$

Actual stress at given pressure and thickness, corroded [Sact]:

$$\begin{aligned}
 &= (P \cdot (D + 1.2 \cdot t \cdot \cos(a))) / (2 \cdot E \cdot t \cdot \cos(a)) \\
 &= (0.5 \cdot (1400.0 + 1.2 \cdot 3.0 \cdot 0.9937)) / (2 \cdot 1.0 \cdot 3.0 \cdot 0.9937) \\
 &= 11.771 \text{ N./mm}^2
 \end{aligned}$$

% Elongation per Table UG-79-1  $(50 \cdot t_{nom} / R_f) \cdot (1 - R_f / R_o)$  0.315 %

Note: Please Check Requirements of Table UHA-44 for Elongation limits.

SA-240 316, Min Metal Temp without impact per UHA-51: -196 °C

Elements Suitable for Internal Pressure.

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

**External Pressure Calculation Results :****External Pressure Calculations:**

From	To	Section Length mm.	Outside Diameter mm.	Corroded Thickness mm.	Factor A	Factor B N./mm <sup>2</sup>
10	Ring	3958	1306	3	0.46719E-04	4.48502
Ring	Ring	1988	1306	3	0.95296E-04	9.14839
Ring	Ring	1993	1306	3	0.95050E-04	9.12477
Ring	Ring	12.9998	1306	3	0.0047922	91.4355
Ring	20	100	1306	3	0.0026	83.9018
20	Ring	1679.94	1306.03	3	0.00011239	10.7893
Ring	30	1752.3	1736.03	3	0.93196E-04	8.94674
30	Ring	No Calc	...	3	No Calc	No Calc
Ring	40	800	1736	3	0.00021156	20.3093
40	Ring	1752.3	1736.06	3	0.92641E-04	8.89351
Ring	Ring	1085.13	1736.06	3	0.00015138	14.5319
Ring	50	0.99994	1406.33	3	0.0041294	89.7316
50	Ring	90	1406	3	0.0029414	85.5356
Ring	Ring	13	1406	3	0.0042047	89.9431
Ring	Ring	1410	1406	3	0.000131	12.5764
Ring	Ring	1400	1406	3	0.00013196	12.6681
Ring	60	1587	1406	3	0.00011601	11.1366
60	Ring	0.99994	1406.02	3	0.004178	89.8686
Ring	Ring	1809.08	1735.86	3	0.90550E-04	8.69271
Ring	70	0.99989	1736.02	3	0.0028755	85.24
70	Ring	737	1736	3	0.00023058	22.1352
Ring	Ring	13	1736	3	0.0028938	85.3229
Ring	80	50	1736	3	0.0028938	85.3229
80	Ring	0.99989	1736.06	3	0.0028434	85.0923
Ring	Ring	1085.13	1736.06	3	0.00015138	14.5319
Ring	90	0.99994	1406.33	3	0.0041294	89.7316
90	Ring	375	1406	3	0.00052804	50.6919
Ring	Ring	1370	1406	3	0.0001349	12.9498
Ring	100	1755	1406	3	0.00010461	10.0421
100	Ring	0.99994	1406.04	3	0.0041551	89.8045
Ring	Ring	1679.26	1406.04	3	0.00010837	10.4036
Ring	110	0.99982	956.263	3	0.0083163	97.0162

**External Pressure Calculations:**

From	To	External Actual T. mm.	External Required T. mm.	External Design Pressure bars	External M.A.W.P. bars
10	Ring	3	2.64879	0.1	0.13736
Ring	Ring	3	1.98494	0.1	0.28018
Ring	Ring	3	1.98708	0.1	0.27946
Ring	Ring	3	0.47257	0.1	2.80031
Ring	20	3	0.57319	0.1	2.56958
20	Ring	3	1.86049	0.1	0.32853
Ring	30	3	2.24961	0.1	0.20495
30	Ring	3	No Calc	0.1	No Calc
Ring	40	3	1.62103	0.1	0.46793
40	Ring	3	2.25778	0.1	0.20299
Ring	Ring	3	1.85593	0.1	0.33169
Ring	50	3	0.51366	0.1	2.52828
50	Ring	3	0.56956	0.1	2.4333
Ring	Ring	3	0.50876	0.1	2.55869
Ring	Ring	3	1.80715	0.1	0.35777
Ring	Ring	3	1.80239	0.1	0.36038
Ring	60	3	1.89031	0.1	0.31681

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60	Ring	3	0.51049	0.1	2.54788
Ring	Ring	3	2.27117	0.1	0.19962
Ring	70	3	0.6303	0.1	1.95727
70	Ring	3	1.56614	0.1	0.51
Ring	Ring	3	0.62816	0.1	1.96585
Ring	80	3	0.62816	0.1	1.96585
80	Ring	3	0.63409	0.1	1.9422
Ring	Ring	3	1.85593	0.1	0.33169
Ring	90	3	0.51366	0.1	2.52828
90	Ring	3	1.04348	0.1	1.44208
Ring	Ring	3	1.78809	0.1	0.3684
Ring	100	3	1.96728	0.1	0.28568
100	Ring	3	0.51198	0.1	2.53866
Ring	Ring	3	1.94504	0.1	0.2941
Ring	110	3	0.34821	0.1	4.03247
Minimum					0.137

**External Pressure Calculations:**

From	To	Actual Length Bet. Stiffeners mm.	Allowable Length Bet. Stiffeners mm.	Ring Inertia Required cm**4	Ring Inertia Available cm**4
10	Ring	3958	5366.66	No Calc	No Calc
Ring	Ring	1988	5367.77	4.74696	17.6358
Ring	Ring	1993	5367.78	3.17821	17.6358
Ring	Ring	12.9998	362.283	1.60148	17.6358
Ring	20	100	3105.85	0.090213	17.6358
20	Ring	1679.94	1679.94	No Calc	No Calc
Ring	30	1752.3	1752.3	2.74012	118.645
30	Ring	No Calc	No Calc	No Calc	No Calc
Ring	40	800	3576.41	1.50003	18.5486
40	Ring	1752.3	1752.3	No Calc	No Calc
Ring	Ring	1085.13	1085.13	5.32031	21.7628
Ring	50	0.99994	0.99994	1.08257	17.7314
50	Ring	90	2782.45	No Calc	No Calc
Ring	Ring	13	332.63	0.1026	17.8679
Ring	Ring	1410	4840.94	1.41706	17.8679
Ring	Ring	1400	4840.92	2.80393	17.8679
Ring	60	1587	4841.33	2.96283	17.8679
60	Ring	0.99994	0.99994	No Calc	No Calc
Ring	Ring	1809.08	1809.08	1.80308	17.8679
Ring	70	0.99989	0.99989	3.39301	18.5482
70	Ring	737	3576.31	No Calc	No Calc
Ring	Ring	13	979.513	1.41118	18.5486
Ring	80	50	1799.98	0.11813	18.5486
80	Ring	0.99989	0.99989	No Calc	No Calc
Ring	Ring	1085.13	1085.13	2.0298	18.381
Ring	90	0.99994	0.99994	1.08257	17.7021
90	Ring	375	4840.98	No Calc	No Calc
Ring	Ring	1370	4840.85	1.73585	17.8679
Ring	100	1755	4841.66	3.11292	17.8679
100	Ring	0.99994	0.99994	No Calc	No Calc
Ring	Ring	1679.26	1679.26	1.67377	21.4269
Ring	110	0.99982	0.99982	0.52653	20.0339

Elements Suitable for External Pressure.

**ASME Code, Section VIII Division 1, 2017****Cylindrical Shell From 10 to Ring:[1 of 1] Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

FileName : Circular Sections

External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

## Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	3958.00	435.33	3.0306	0.0000467	4.49
EMAP = (4*B) / (3*(D/t)) = (4*4.485) / (3*435.3333) = 0.1374 bars						

## Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.649	1306.00	3958.00	493.06	3.0306	0.0000385	3.70
EMAP = (4*B) / (3*(D/t)) = (4*3.6983) / (3*493.0556) = 0.1 bars						

## Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	5366.66	435.33	4.1092	0.0000340	3.27
EMAP = (4*B) / (3*(D/t)) = (4*3.268) / (3*435.3333) = 0.1001 bars						

**Cylindrical Shell From Ring:[1 of 1] to Ring:[2 of 2] Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

## Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	1988.00	435.33	1.5222	0.0000953	9.15
EMAP = (4*B) / (3*(D/t)) = (4*9.1484) / (3*435.3333) = 0.2802 bars						

## Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.985	1306.00	1988.00	657.96	1.5222	0.0000514	4.94
EMAP = (4*B) / (3*(D/t)) = (4*4.9351) / (3*657.9558) = 0.1 bars						

## Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	5367.77	435.33	4.1101	0.0000340	3.27
EMAP = (4*B) / (3*(D/t)) = (4*3.2673) / (3*435.3333) = 0.1001 bars						

**Cylindrical Shell From Ring:[2 of 2] to Ring:[3 of 3] Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

## Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	1993.00	435.33	1.5260	0.0000951	9.12
EMAP = (4*B) / (3*(D/t)) = (4*9.1248) / (3*435.3333) = 0.2795 bars						

## Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.987	1306.00	1993.00	657.25	1.5260	0.0000514	4.93
EMAP = (4*B) / (3*(D/t)) = (4*4.9298) / (3*657.2463) = 0.1 bars						

## Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1306.00	5367.78	435.33	4.1101	0.0000340	3.27
EMAP = (4*B) / (3*(D/t)) = (4*3.2673) / (3*435.3333) = 0.1001 bars						

**Cone From Ring 16 to 30 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

## Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.983	1736.03	1752.30	582.01	1.0094	0.0000932	8.95
EMAP = (4*B) / (3*(D/t)) = (4*8.9467) / (3*582.0123) = 0.2049 bars						

Note: The cone thickness used in the calculation has been modified  
per UG-33(f),  $t_e = t * \cos(\alpha)$ .

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

## Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.237	1736.03	1752.30	776.15	1.0094	0.0000606	5.82

EMAP =  $(4*B)/(3*(D/t)) = (4*5.8216)/(3*776.1512) = 0.1$  bars

Note: The cone thickness used in the calculation has been modified  
per UG-33(f),  $t_e = t * \cos(\alpha)$ .

**Cylindrical Shell From Ring5 to 40 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

## Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1736.00	800.00	578.67	0.4608	0.0002116	20.31

EMAP =  $(4*B)/(3*(D/t)) = (4*20.3093)/(3*578.6667) = 0.4679$  bars

## Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.621	1736.00	800.00	1070.92	0.4608	0.0000837	8.03

EMAP =  $(4*B)/(3*(D/t)) = (4*8.0326)/(3*1070.9225) = 0.1$  bars

## Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1736.00	3576.41	578.67	2.0601	0.0000452	4.34

EMAP =  $(4*B)/(3*(D/t)) = (4*4.3427)/(3*578.6667) = 0.1001$  bars

**Cone From Ring 21 to Ring:17 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

## Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.972	1736.06	1085.13	584.13	0.6251	0.0001514	14.53

EMAP =  $(4*B)/(3*(D/t)) = (4*14.5319)/(3*584.1302) = 0.3317$  bars

Note: The cone thickness used in the calculation has been modified  
per UG-33(f),  $t_e = t * \cos(\alpha)$ .

## Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.839	1736.06	1085.13	944.21	0.6251	0.0000738	7.08

EMAP =  $(4*B)/(3*(D/t)) = (4*7.0822)/(3*944.2127) = 0.1$  bars

Note: The cone thickness used in the calculation has been modified  
per UG-33(f),  $t_e = t * \cos(\alpha)$ .

**Cone From Ring:17 to 50 Ext. Chart: HA-2 at 50 °C****Cylindrical Shell From Ring 7 to Ring 8 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

## Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	1410.00	468.67	1.0028	0.0001310	12.58

EMAP =  $(4*B)/(3*(D/t)) = (4*12.5764)/(3*468.6667) = 0.3578$  bars

## Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.807	1406.00	1410.00	778.02	1.0028	0.0000608	5.84

EMAP =  $(4*B)/(3*(D/t)) = (4*5.8358)/(3*778.0202) = 0.1$  bars



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## Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4840.94	468.67	3.4431	0.0000366	3.52

EMAP =  $(4*B)/(3*(D/t)) = (4*3.5182)/(3*468.6667) = 0.1001$  bars

**Cylindrical Shell From Ring 8 to Ring 9 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

## Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	1400.00	468.67	0.9957	0.0001320	12.67

EMAP =  $(4*B)/(3*(D/t)) = (4*12.6681)/(3*468.6667) = 0.3604$  bars

## Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.802	1406.00	1400.00	780.07	0.9957	0.0000610	5.85

EMAP =  $(4*B)/(3*(D/t)) = (4*5.8512)/(3*780.0737) = 0.1$  bars

## Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4840.92	468.67	3.4430	0.0000366	3.52

EMAP =  $(4*B)/(3*(D/t)) = (4*3.5182)/(3*468.6667) = 0.1001$  bars

**Cylindrical Shell From Ring 9 to 60 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

## Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	1587.00	468.67	1.1287	0.0001160	11.14

EMAP =  $(4*B)/(3*(D/t)) = (4*11.1366)/(3*468.6667) = 0.3168$  bars

## Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.890	1406.00	1587.00	743.79	1.1287	0.0000581	5.58

EMAP =  $(4*B)/(3*(D/t)) = (4*5.5789)/(3*743.7935) = 0.1$  bars

## Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4841.33	468.67	3.4433	0.0000366	3.52

EMAP =  $(4*B)/(3*(D/t)) = (4*3.5179)/(3*468.6667) = 0.1001$  bars

**Cone From Ring 18 to Ring 23 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

## Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.990	1735.86	1809.08	580.58	1.0422	0.0000905	8.69

EMAP =  $(4*B)/(3*(D/t)) = (4*8.6927)/(3*580.5842) = 0.1996$  bars

Note: The cone thickness used in the calculation has been modified  
per UG-33(f),  $t_e = t * \cos(\alpha)$ .

## Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.263	1735.86	1809.08	766.90	1.0422	0.0000599	5.75

EMAP =  $(4*B)/(3*(D/t)) = (4*5.7522)/(3*766.8951) = 0.1$  bars

Note: The cone thickness used in the calculation has been modified  
per UG-33(f),  $t_e = t * \cos(\alpha)$ .

**Cone From Ring 23 to 70 Ext. Chart: HA-2 at 50 °C**

FileName : Circular Sections

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**Cylindrical Shell From 70 to Ring 10 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1736.00	737.00	578.67	0.4245	0.0002306	22.14

EMAP =  $(4*B)/(3*(D/t)) = (4*22.1352)/(3*578.6667) = 0.51$  bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.566	1736.00	737.00	1108.46	0.4245	0.0000866	8.31

EMAP =  $(4*B)/(3*(D/t)) = (4*8.3142)/(3*1108.459) = 0.1$  bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1736.00	3576.31	578.67	2.0601	0.0000452	4.34

EMAP =  $(4*B)/(3*(D/t)) = (4*4.3428)/(3*578.6667) = 0.1001$  bars

**Cone From Ring 24 to Ring 26 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.972	1736.06	1085.13	584.13	0.6251	0.0001514	14.53

EMAP =  $(4*B)/(3*(D/t)) = (4*14.5319)/(3*584.1302) = 0.3317$  bars

Note: The cone thickness used in the calculation has been modified  
per UG-33(f),  $t_e = t * \cos(\alpha)$ .

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.839	1736.06	1085.13	944.21	0.6251	0.0000738	7.08

EMAP =  $(4*B)/(3*(D/t)) = (4*7.0822)/(3*944.2127) = 0.1$  bars

Note: The cone thickness used in the calculation has been modified  
per UG-33(f),  $t_e = t * \cos(\alpha)$ .

**Cone From Ring 26 to 90 Ext. Chart: HA-2 at 50 °C****Cylindrical Shell From 90 to Ring 12 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	375.00	468.67	0.2667	0.0005280	50.69

EMAP =  $(4*B)/(3*(D/t)) = (4*50.6919)/(3*468.6667) = 1.4421$  bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.043	1406.00	375.00	1347.41	0.2667	0.0001053	10.11

EMAP =  $(4*B)/(3*(D/t)) = (4*10.1068)/(3*1347.4142) = 0.1$  bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4840.98	468.67	3.4431	0.0000366	3.52

EMAP =  $(4*B)/(3*(D/t)) = (4*3.5182)/(3*468.6667) = 0.1001$  bars

**Cylindrical Shell From Ring 12 to Ring 13] Ext. Chart: HA-2 at 50 °C**

FileName : Circular Sections -----

External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	1370.00	468.67	0.9744	0.0001349	12.95

EMAP =  $(4*B)/(3*(D/t)) = (4*12.9498)/(3*468.6667) = 0.3684$  bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.788	1406.00	1370.00	786.31	0.9744	0.0000614	5.90

EMAP =  $(4*B)/(3*(D/t)) = (4*5.898)/(3*786.3148) = 0.1$  bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4840.85	468.67	3.4430	0.0000366	3.52

EMAP =  $(4*B)/(3*(D/t)) = (4*3.5183)/(3*468.6667) = 0.1001$  bars

**Cylindrical Shell From Ring 13] to 100 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	1755.00	468.67	1.2482	0.0001046	10.04

EMAP =  $(4*B)/(3*(D/t)) = (4*10.0421)/(3*468.6667) = 0.2857$  bars

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.967	1406.00	1755.00	714.69	1.2482	0.0000558	5.36

EMAP =  $(4*B)/(3*(D/t)) = (4*5.3608)/(3*714.6927) = 0.1$  bars

Results for Maximum Stiffened Length (Slen):

Tca	OD	SLEN	D/t	L/D	Factor A	B
3.000	1406.00	4841.66	468.67	3.4436	0.0000366	3.52

EMAP =  $(4*B)/(3*(D/t)) = (4*3.5177)/(3*468.6667) = 0.1001$  bars

**Cone From Ring 28 to Ring 22 Ext. Chart: HA-2 at 50 °C**

Elastic Modulus from Chart: HA-2 at 50 °C : 0.192E+09 KPa.

Results for Maximum Allowable External Pressure (MAEP):

Tca	OD	SLEN	D/t	L/D	Factor A	B
2.981	1406.04	1679.26	471.64	1.1943	0.0001084	10.40

EMAP =  $(4*B)/(3*(D/t)) = (4*10.4036)/(3*471.6358) = 0.2941$  bars

Note: The cone thickness used in the calculation has been modified  
per UG-33(f),  $t_e = t * \cos(\alpha)$ .

Results for Required Thickness (Tca):

Tca	OD	SLEN	D/t	L/D	Factor A	B
1.933	1406.04	1679.26	727.44	1.1943	0.0000568	5.46

EMAP =  $(4*B)/(3*(D/t)) = (4*5.4564)/(3*727.4449) = 0.1$  bars

Note: The cone thickness used in the calculation has been modified  
per UG-33(f),  $t_e = t * \cos(\alpha)$ .

**Cone From Ring 22 to the end: Ext. Chart: HA-2 at 50 °C**

**Stiffening Ring Calcs for : Ring:[1 of 1] , SA-240 316 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 69 mm.

Area (cm²)	Distance (mm.)	Area*Dist
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Shell:	2.066	1.5000	3.098
Ring :	4.700	26.5000	124.550
Total:	6.766		127.648

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.015	17.3673	623.030
Ring :	8.652	-7.6327	273.815
Total:	8.667		896.845

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell	Breq	3.10 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000323

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1306.0^2 * 2973.0002(3.0 + 4.7/2973.0002)0.000032) / 10.9$$

$$= 5 \text{ cm}^4$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	3.03	Kgf/mm.
The Radial Shear Load	V	39.59	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3587.38	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.81	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.59	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	45.03	Kgf/mm.
The Combined Weld Load SRSS of VQ/I and Pext*Slen		3.14	Kgf/mm.

**Stiffening Ring Calcs for : Ring:[2 of 2], SA-240 316 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 69 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.066	1.5000	3.098
Ring :	4.700	26.5000	124.550
Total:	6.766		127.648

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.015	17.3673	623.030
Ring :	8.652	-7.6327	273.815
Total:	8.667		896.845

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell	Breq	3.03 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000315

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1306.0^2 * 1990.5(3.0 + 4.7/1990.5)0.000032) / 10.9$$

$$= 3 \text{ cm}^4$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
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Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	2.03	Kgf/mm.
The Radial Shear Load	V	26.51	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3587.38	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.54	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.59	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	45.03	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	2.10	Kgf/mm.

**Stiffening Ring Calcs for : Ring:[3 of 3], SA-240 316 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 69 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.066	1.5000	3.098
Ring :	4.700	26.5000	124.550
Total:	6.766		127.648

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.015	17.3673	623.030
Ring :	8.652	-7.6327	273.815
Total:	8.667		896.845

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell	Breq	2.82	N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000294	

**Required Moment of Inertia, Ring plus Shell:**

$$= ( OD^2 * Slen(Tca + Aring/Slen)Areq )/10.9$$

$$= (1306.0^2 * 1003.0(3.0 + 4.7/1003.0)0.0000294)/10.9$$

$$= 2 \text{ cm**4}$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	1.02	Kgf/mm.
The Radial Shear Load	V	13.36	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3587.38	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.27	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.59	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	45.03	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	1.06	Kgf/mm.

**Stiffening Ring Calcs for : Ring:[4 of 4], SA-240 316 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 69 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.066	1.5000	3.098
Ring :	4.700	26.5000	124.550
Total:	6.766		127.648

Centroid of Ring plus Shell: 19 mm.

Inertia	Distance	A*Dist <sup>2</sup>
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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Shell:	0.015	17.3673	623.030
Ring :	8.652	-7.6327	273.815
Total:	8.667		896.845

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell	Breq	0.87 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000090

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1306.0^2 * 56.5(3.0 + 4.7/56.5)0.000009) / 10.9$$

$$= 0 \text{ cm}^4$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.06	Kgf/mm.
The Radial Shear Load	V	0.75	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3587.38	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.02	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.59	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	45.03	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.06	Kgf/mm.

**Stiffening Ring Calcs for : Ring 16 , SA-240 304 , Bar Ring: 97 x 10 mm.**

Effective Length of Shell: 69 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.066	1.5000	3.098
Ring :	9.700	51.5000	499.550
Total:	11.766		502.648

Centroid of Ring plus Shell: 43 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.015	41.2219	3509.954
Ring :	76.056	-8.7781	747.439
Total:	76.071		4257.393

Available Moment of Inertia, Ring plus Shell: 119 cm\*\*4

Required Stress in Ring plus Shell	Breq	2.75 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000286

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1306.0^2 * 1716.1235(3.0 + 9.7/1716.1235)0.000029) / 10.9$$

$$= 3 \text{ cm}^4$$

**Stiffening Ring Calcs for : Ring5 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 79 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.381	1.5000	3.572
Ring :	4.700	26.5000	124.550
Total:	7.081		128.122

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

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Centroid of Ring plus Shell: 18 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.018	16.5926	655.656
Ring :	8.652	-8.4075	332.221
Total:	8.670		987.878

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Available Moment of Inertia, Ring plus Shell: 19 cm\*\*4

Required Stress in Ring plus Shell	Breq	3.12 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000325

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1736.0^2 * 400.0(3.0 + 4.7/400.0)0.000032) / 10.9$$

$$= 2 \text{ cm**4}$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.41	Kgf/mm.
The Radial Shear Load	V	7.08	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3951.51	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.15	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.43	Kgf/mm.

**Stiffening Ring Calcs for : Ring 21 , SA-240 304 , Bar Ring: 50 x 10 mm.**

Effective Length of Shell: 79 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.381	1.5000	3.572
Ring :	5.000	28.0000	140.000
Total:	7.381		143.572

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Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.018	17.9503	767.347
Ring :	10.417	-8.5497	365.487
Total:	10.434		1132.833

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Available Moment of Inertia, Ring plus Shell: 22 cm\*\*4

Required Stress in Ring plus Shell	Breq	3.88 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000405

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1736.0^2 * 1418.7195(3.0 + 5.0/1418.7195)0.00004) / 10.9$$

$$= 5 \text{ cm**4}$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	1.45	Kgf/mm.

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

The Radial Shear Load	V	25.12	Kgf
The First Moment of the Area ( Ring + Shell )	Q	4274.85	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.49	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	1.53	Kgf/mm.

**Stiffening Ring Calcs for : Ring:17 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 71 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.686	26.4300	123.851
Total:	6.829		127.066

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.016	17.1057	627.176
Ring :	8.575	-7.8243	286.878
Total:	8.591		914.053

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell	Breq	2.73 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000284

**Required Moment of Inertia, Ring plus Shell:**

$$= ( OD^2 * Slen(Tca + Aring/Slen)Areq ) / 10.9$$

$$= (1406.275^2 * 543.0673(3.0 + 4.686/543.0673)0.0000284) / 10.9$$

$$= 1 \text{ cm}^{*4}$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.55	Kgf/mm.
The Radial Shear Load	V	7.79	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3666.48	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.16	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.58	Kgf/mm.

**Stiffening Ring Calcs for : Ring 6 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 71 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132



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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Total: 8.668 919.992  
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 Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4  
 Required Stress in Ring plus Shell Breq 0.87 N./mm<sup>2</sup>  
 Required Strain in Ring plus Shell Areq 0.0000091

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 51.5(3.0 + 4.7/51.5)0.000009) / 10.9$$

$$= 0 \text{ cm**4}$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.  
 Stiffening Ring Attachment Style CONTINUOUS  
 Location of Stiffening Ring EXTERNAL  
 Radial Pressure Load Pext\*Slen 0.05 Kgf/mm.  
 The Radial Shear Load V 0.74 Kgf  
 The First Moment of the Area ( Ring + Shell ) Q 3679.97 mm.<sup>3</sup>  
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.02 Kgf/mm.  
 The Weld Allowable Stress 0.55\*S 73.09 N./mm<sup>2</sup>  
 Minimum Weld Leg Size Min( 6mm, t, tw ) Wldmin 3.00 mm.  
 The Weld Allowable Load 2\* WLeg\*0.55\*S 44.72 Kgf/mm.  
 The Combined Weld Load SRSS of VQ/I and Pext\*Slen 0.05 Kgf/mm.

**Stiffening Ring Calcs for : Ring 7 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 71 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell Breq 2.88 N./mm<sup>2</sup>  
 Required Strain in Ring plus Shell Areq 0.0000300

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 711.5(3.0 + 4.7/711.5)0.00003) / 10.9$$

$$= 1 \text{ cm**4}$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.  
 Stiffening Ring Attachment Style CONTINUOUS  
 Location of Stiffening Ring EXTERNAL  
 Radial Pressure Load Pext\*Slen 0.73 Kgf/mm.  
 The Radial Shear Load V 10.20 Kgf  
 The First Moment of the Area ( Ring + Shell ) Q 3679.97 mm.<sup>3</sup>  
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.21 Kgf/mm.  
 The Weld Allowable Stress 0.55\*S 73.09 N./mm<sup>2</sup>  
 Minimum Weld Leg Size Min( 6mm, t, tw ) Wldmin 3.00 mm.  
 The Weld Allowable Load 2\* WLeg\*0.55\*S 44.72 Kgf/mm.  
 The Combined Weld Load SRSS of VQ/I and Pext\*Slen 0.76 Kgf/mm.

FileName : Circular Sections

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**Stiffening Ring Calcs for : Ring 8 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 71 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell Breq 3.16 N./mm<sup>2</sup>  
 Required Strain in Ring plus Shell Areq 0.0000330

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 1405.0(3.0 + 4.7/1405.0)0.000033) / 10.9$$

$$= 3 \text{ cm}^{**4}$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.  
 Stiffening Ring Attachment Style CONTINUOUS  
 Location of Stiffening Ring EXTERNAL  
 Radial Pressure Load Pext\*Slen 1.43 Kg/mm.  
 The Radial Shear Load V 20.14 Kg  
 The First Moment of the Area ( Ring + Shell ) Q 3679.97 mm.<sup>3</sup>  
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.41 Kg/mm.  
 The Weld Allowable Stress 0.55\*S 73.09 N./mm<sup>2</sup>  
 Minimum Weld Leg Size Min( 6mm, t, tw ) Wldmin 3.00 mm.  
 The Weld Allowable Load 2\* Wleg\*0.55\*S 44.72 Kg/mm.  
 The Combined Weld Load SRSS of VQ/I and Pext\*Slen 1.49 Kg/mm.

**Stiffening Ring Calcs for : Ring 9 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 71 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell Breq 3.18 N./mm<sup>2</sup>  
 Required Strain in Ring plus Shell Areq 0.0000330

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#### Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 1493.5001(3.0 + 4.7/1493.5001)0.000033) / 10.9$$

$$= 3 \text{ cm}^4$$

#### Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	1.52	Kgf/mm.
The Radial Shear Load	V	21.41	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3679.97	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.44	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	1.59	Kgf/mm.

#### Stiffening Ring Calcs for : Ring 18 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 71 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm<sup>4</sup>

Required Stress in Ring plus Shell Breq 3.00 N./mm<sup>2</sup>  
 Required Strain in Ring plus Shell Areq 0.0000312

#### Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 905.0399(3.0 + 4.7/905.0399)0.000031) / 10.9$$

$$= 2 \text{ cm}^4$$

#### Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	0.92	Kgf/mm.
The Radial Shear Load	V	12.98	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3679.97	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.27	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.96	Kgf/mm.

#### Stiffening Ring Calcs for : Ring 23 , SA-240 304 , Bar Ring: 47 x 10 mm.

Effective Length of Shell: 79 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
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Shell:	2.381	1.5000	3.572
Ring :	4.700	26.5000	124.550
Total:	7.081		128.122

Centroid of Ring plus Shell: 18 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.018	16.5928	655.646
Ring :	8.652	-8.4072	332.200
Total:	8.670		987.847

Available Moment of Inertia, Ring plus Shell: 19 cm\*\*4

Required Stress in Ring plus Shell	Breq	3.70 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000385

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1735.8351^2 * 905.0399(3.0 + 4.7/905.0399)0.000039) / 10.9$$

$$= 3 \text{ cm}^4$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.92	Kgf/mm.
The Radial Shear Load	V	16.02	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3951.38	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.34	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load SRSS of VQ/I and Pext*Slen		0.98	Kgf/mm.

**Stiffening Ring Calcs for : Ring 10 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 79 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.381	1.5000	3.572
Ring :	4.700	26.5000	124.550
Total:	7.081		128.122

Centroid of Ring plus Shell: 18 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.018	16.5926	655.656
Ring :	8.652	-8.4075	332.221
Total:	8.670		987.878

Available Moment of Inertia, Ring plus Shell: 19 cm\*\*4

Required Stress in Ring plus Shell	Breq	3.06 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000320

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1736.0^2 * 375.0(3.0 + 4.7/375.0)0.000032) / 10.9$$

$$= 1 \text{ cm}^4$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
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Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	0.38	Kgf/mm.
The Radial Shear Load	V	6.64	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3951.51	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.14	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.41	Kgf/mm.

**Stiffening Ring Calcs for : Ring 11 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 79 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.381	1.5000	3.572
Ring :	4.700	26.5000	124.550
Total:	7.081		128.122

Centroid of Ring plus Shell: 18 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.018	16.5926	655.656
Ring :	8.652	-8.4075	332.221
Total:	8.670		987.878

Available Moment of Inertia, Ring plus Shell: 19 cm\*\*4

Required Stress in Ring plus Shell Breq 0.73 N./mm<sup>2</sup>  
 Required Strain in Ring plus Shell Areq 0.0000076

**Required Moment of Inertia, Ring plus Shell:**

$$= ( OD^2 * Slen(Tca + Aring/Slen)Areq )/10.9$$

$$= (1736.0^2 * 31.5(3.0 + 4.7/31.5)0.000008)/10.9$$

$$= 0 \text{ cm**4}$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	0.03	Kgf/mm.
The Radial Shear Load	V	0.56	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3951.51	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.01	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.03	Kgf/mm.

**Stiffening Ring Calcs for : Ring 24 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 79 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.381	1.5000	3.572
Ring :	4.684	26.4175	123.727
Total:	7.065		127.299

Centroid of Ring plus Shell: 18 mm.

Inertia	Distance	A*Dist <sup>2</sup>
---------	----------	---------------------

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Shell:	0.018	16.5182	649.795
Ring :	8.561	-8.3993	330.411
Total:	8.579		980.207

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell	Breq	3.37 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000350

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1736.0^2 * 543.0673(3.0 + 4.6835/543.0673)0.000035) / 10.9$$

$$= 2 \text{ cm}^4$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.55	Kgf/mm.
The Radial Shear Load	V	9.61	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3933.81	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.21	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* Wleg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.59	Kgf/mm.

**Stiffening Ring Calcs for : Ring 26 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 71 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.683	26.4150	123.701
Total:	6.826		126.916

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.016	17.0919	626.168
Ring :	8.558	-7.8231	286.600
Total:	8.574		912.769

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell	Breq	2.73 N./mm <sup>2</sup>
Required Strain in Ring plus Shell	Areq	0.0000284

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.275^2 * 543.0673(3.0 + 4.683/543.0673)0.000028) / 10.9$$

$$= 1 \text{ cm}^4$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style		CONTINUOUS	
Location of Stiffening Ring		EXTERNAL	
Radial Pressure Load	Pext*Slen	0.55	Kgf/mm.
The Radial Shear Load	V	7.79	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3663.53	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.16	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Minimum Weld Leg Size Min( 6mm, t, tw ) Wldmin 3.00 mm.  
 The Weld Allowable Load 2\* WLeg\*0.55\*S 44.72 Kgf/mm.  
 The Combined Weld Load SRSS of VQ/I and Pext\*Slen 0.58 Kgf/mm.

**Stiffening Ring Calcs for : Ring 12 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 71 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

Required Stress in Ring plus Shell Breq 2.98 N./mm<sup>2</sup>  
 Required Strain in Ring plus Shell Areq 0.0000310

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 872.5(3.0 + 4.7/872.5)0.000031) / 10.9$$

$$= 2 \text{ cm}^{**4}$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.  
 Stiffening Ring Attachment Style CONTINUOUS  
 Location of Stiffening Ring EXTERNAL  
 Radial Pressure Load Pext\*Slen 0.89 Kgf/mm.  
 The Radial Shear Load V 12.51 Kgf  
 The First Moment of the Area ( Ring + Shell ) Q 3679.97 mm.<sup>3</sup>  
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.26 Kgf/mm.  
 The Weld Allowable Stress 0.55\*S 73.09 N./mm<sup>2</sup>  
 Minimum Weld Leg Size Min( 6mm, t, tw ) Wldmin 3.00 mm.  
 The Weld Allowable Load 2\* WLeg\*0.55\*S 44.72 Kgf/mm.  
 The Combined Weld Load SRSS of VQ/I and Pext\*Slen 0.93 Kgf/mm.

**Stiffening Ring Calcs for : Ring 131 , SA-240 304 , Bar Ring: 47 x 10 mm.**

Effective Length of Shell: 71 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	4.700	26.5000	124.550
Total:	6.843		127.765

Centroid of Ring plus Shell: 19 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.016	17.1703	631.861
Ring :	8.652	-7.8297	288.132
Total:	8.668		919.992

Available Moment of Inertia, Ring plus Shell: 18 cm\*\*4

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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

Required Stress in Ring plus Shell Breq 3.19 N./mm<sup>2</sup>  
 Required Strain in Ring plus Shell Areq 0.0000333

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 1562.5(3.0 + 4.7/1562.5)0.000033) / 10.9$$

$$= 3 \text{ cm}^4$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.  
 Stiffening Ring Attachment Style CONTINUOUS  
 Location of Stiffening Ring EXTERNAL  
 Radial Pressure Load Pext\*Slen 1.59 Kgf/mm.  
 The Radial Shear Load V 22.40 Kgf  
 The First Moment of the Area ( Ring + Shell ) Q 3679.97 mm.<sup>3</sup>  
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.46 Kgf/mm.  
 The Weld Allowable Stress 0.55\*S 73.09 N./mm<sup>2</sup>  
 Minimum Weld Leg Size Min( 6mm, t, tw ) Wldmin 3.00 mm.  
 The Weld Allowable Load 2\* WLeg\*0.55\*S 44.72 Kgf/mm.  
 The Combined Weld Load SRSS of VQ/I and Pext\*Slen 1.66 Kgf/mm.

**Stiffening Ring Calcs for : Ring 28 , SA-240 304 , Bar Ring: 50 x 10 mm.**

Effective Length of Shell: 71 mm.

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	2.143	1.5000	3.215
Ring :	5.042	28.2100	142.235
Total:	7.185		145.450

Centroid of Ring plus Shell: 20 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.016	18.7429	752.904
Ring :	10.681	-7.9671	320.040
Total:	10.697		1072.945

Available Moment of Inertia, Ring plus Shell: 21 cm<sup>4</sup>

Required Stress in Ring plus Shell Breq 2.93 N./mm<sup>2</sup>  
 Required Strain in Ring plus Shell Areq 0.0000305

**Required Moment of Inertia, Ring plus Shell:**

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (1406.0^2 * 840.1315(3.0 + 5.042/840.1315)0.000031) / 10.9$$

$$= 2 \text{ cm}^4$$

**Results for Stiffening Ring Weld Calculations per UG-30:**

Given Stiffening Ring Fillet Weld Size Wleg 3.000 mm.  
 Stiffening Ring Attachment Style CONTINUOUS  
 Location of Stiffening Ring EXTERNAL  
 Radial Pressure Load Pext\*Slen 0.86 Kgf/mm.  
 The Radial Shear Load V 12.05 Kgf  
 The First Moment of the Area ( Ring + Shell ) Q 4017.02 mm.<sup>3</sup>  
 Weld Shear Flow due to Rad. Shear Load VQ/I 0.23 Kgf/mm.  
 The Weld Allowable Stress 0.55\*S 73.09 N./mm<sup>2</sup>  
 Minimum Weld Leg Size Min( 6mm, t, tw ) Wldmin 3.00 mm.  
 The Weld Allowable Load 2\* WLeg\*0.55\*S 44.72 Kgf/mm.  
 The Combined Weld Load SRSS of VQ/I and Pext\*Slen 0.89 Kgf/mm.

**Stiffening Ring Calcs for : Ring 22 , SA-240 304 , Bar Ring: 50 x 10 mm.**

Effective Length of Shell: 59 mm.



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External Pressure Calculations: Step: 14 4:14pm Feb 20,2026

	Area (cm <sup>2</sup> )	Distance (mm.)	Area*Dist
Shell:	1.767	1.5000	2.651
Ring :	5.042	28.2113	142.248
Total:	6.810		144.899
Centroid of Ring plus Shell:			21 mm.

	Inertia	Distance	A*Dist <sup>2</sup>
Shell:	0.013	19.7783	691.404
Ring :	10.683	-6.9330	242.361
Total:	10.696		933.765
Available Moment of Inertia, Ring plus Shell:			20 cm**4

Required Stress in Ring plus Shell Breq 1.99 N./mm<sup>2</sup>  
 Required Strain in Ring plus Shell Areq 0.0000208

#### Required Moment of Inertia, Ring plus Shell:

$$= (OD^2 * Slen(Tca + Aring/Slen)Areq) / 10.9$$

$$= (956.225^2 * 840.1315(3.0 + 5.0423/840.1315)0.000021) / 10.9$$

$$= 1 \text{ cm**4}$$

#### Results for Stiffening Ring Weld Calculations per UG-30:

Given Stiffening Ring Fillet Weld Size	Wleg	3.000	mm.
Stiffening Ring Attachment Style	CONTINUOUS		
Location of Stiffening Ring	EXTERNAL		
Radial Pressure Load	Pext*Slen	0.86	Kgf/mm.
The Radial Shear Load	V	8.19	Kgf
The First Moment of the Area ( Ring + Shell )	Q	3495.77	mm. <sup>3</sup>
Weld Shear Flow due to Rad. Shear Load	VQ/I	0.14	Kgf/mm.
The Weld Allowable Stress	0.55*S	73.09	N./mm <sup>2</sup>
Minimum Weld Leg Size Min( 6mm, t, tw )	Wldmin	3.00	mm.
The Weld Allowable Load	2* WLeg*0.55*S	44.72	Kgf/mm.
The Combined Weld Load	SRSS of VQ/I and Pext*Slen	0.87	Kgf/mm.

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Element and Detail Weights: Step: 15 4:14pm Feb 20,2026

**Element and Detail Weights:**

From	To	Element Metal Wgt. kg.	Element ID Volume Cm.	Corroded Metal Wgt. kg.	Corroded ID Volume Cm.	Extra due Misc % kg.
10	20	793.769	10689519	793.769	10689519	...
20	30	231.016	3630184	231.016	3630184	...
30	40	104.889	1880831	104.889	1880831	...
40	50	143.693	2317308	143.693	2317308	...
50	60	477.657	6928454	477.657	6928454	...
60	70	238.06	3862180	238.06	3862180	...
70	80	104.889	1880831	104.889	1880831	...
80	90	143.693	2317308	143.693	2317308	...
90	100	371.509	5388796	371.509	5388796	...
100	110	179.371	2195581	179.371	2195581	...
Total		2788	41090996.00	2788	41090996.00	0

**Weight of Details:**

From	Type	Weight of Detail kg.	X Offset, Dtl. Cent. mm.	Y Offset, Dtl. Cent. mm.	Description
10	Ring	16.0367	...	3958	Ring:[1 of 1]
10	Ring	16.0367	...	5946	Ring:[2 of 2]
10	Ring	16.0367	...	7939	Ring:[3 of 3]
10	Ring	16.0367	...	7952	Ring:[4 of 4]
20	Ring	34.3201	...	...	Ring 16
30	Ring	21.1334	...	...	Ring5
40	Ring	17.1723	...	1199	Ring:17
40	Ring	22.5201	...	...	Ring 21
50	Ring	17.222	...	90	Ring 6
50	Ring	17.222	...	103	Ring 7
50	Ring	17.222	...	1513	Ring 8
50	Ring	17.222	...	2913	Ring 9
60	Ring	17.222	...	...	Ring 18
60	Ring	21.1315	...	1999	Ring 23
70	Ring	21.1334	...	737	Ring 10
70	Ring	21.1334	...	750	Ring 11
80	Ring	21.0573	...	...	Ring 24
80	Ring	17.161	...	1199	Ring 26
90	Ring	17.222	...	375	Ring 12
90	Ring	17.222	...	1745	Ring 13]
100	Ring	12.8003	...	1999	Ring 22
100	Ring	18.5186	...	...	Ring 28

**Total Weight of Each Detail Type:**

Stiffeners	412.8
-----	
Sum of the Detail Weights	412.8 kg.

**Weight Summation Results: (kg.)**

	Fabricated	Shop Test	Shipping	Erected	Empty	Operating
-----						
Main Elements	2788.5	2788.5	2788.5	2788.5	2788.5	2788.5
Stif. Rings	412.8	412.8	412.8	412.8	412.8	412.8
Test Liquid	...	41065.9	...	...	...	...
-----						
Totals	3201.3	44267.2	3201.3	3201.3	3201.3	3201.3

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Element and Detail Weights: Step: 15 4:14pm Feb 20,2026

**Weight Summary:**

Fabricated Wt.	- Bare Weight without Removable Internals	3201.3 kg.
Shop Test Wt.	- Fabricated Weight + Water ( Full )	44267.2 kg.
Shipping Wt.	- Fab. Weight + removable Intls.+ Shipping App.	3201.3 kg.
Erected Wt.	- Fab. Wt + or - loose items (trays,platforms etc.)	3201.3 kg.
Ope. Wt. no Liq	- Fab. Weight + Internals. + Details + Weights	3201.3 kg.
Operating Wt.	- Empty Weight + Operating Liq. Uncorroded	3201.3 kg.
Field Test Wt.	- Empty Weight + Water (Full)	44267.2 kg.
Mass of the Vessel	1/3 of the Vertical Vessel	1107.8 kg.

**Outside Surface Areas of Elements:**

From	To	Surface Area cm <sup>2</sup>
10	20	330367
20	30	96120.1
30	40	43630.4
40	50	59784.7
50	60	198769
60	70	99045.5
70	80	43630.4
80	90	59784.7
90	100	154598
100	110	74674.9
Total		1160404.250 cm <sup>2</sup>

**Element and Detail Weights:**

From	To	Total Ele. Empty Wgt. kg.	Total. Ele. Oper. Wgt. kg.	Total. Ele. Hydro. Wgt. kg.	Total Dtl. Offset Mom. Kg-m.	Oper. Wgt. No Liquid kg.
10	20	857.916	857.916	857.916	...	857.916
20	30	265.336	265.336	265.336	...	265.336
30	40	126.022	126.022	126.022	...	126.022
40	50	183.386	183.386	183.386	...	183.386
50	60	546.545	546.545	546.545	...	546.545
60	70	276.413	276.413	276.413	...	276.413
70	80	147.156	147.156	147.156	...	147.156
80	90	181.912	181.912	181.912	...	181.912
90	100	405.953	405.953	405.953	...	405.953
100	110	210.689	210.689	210.689	...	210.689

**Cumulative Vessel Weight**

From	To	Cumulative Ope Wgt. No Liquid kg.	Cumulative Oper. Wgt. kg.	Cumulative Hydro. Wgt. kg.
10	20	3201.33	3201.33	3201.33
20	30	2343.41	2343.41	2343.41
30	40	2078.08	2078.08	2078.08
40	50	1952.05	1952.05	1952.05
50	60	1768.67	1768.67	1768.67
60	70	1222.12	1222.12	1222.12
70	80	945.71	945.71	945.71
80	90	798.554	798.554	798.554
90	100	616.643	616.643	616.643
100	110	210.689	210.689	210.689

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Element and Detail Weights: Step: 15 4:14pm Feb 20,2026

Note: The cumulative operating weights no liquid in the column above are the cumulative operating weights minus the operating liquid weight minus any weights absent in the empty condition.

#### Cumulative Vessel Moment

From	To	Cumulative Empty Mom. Kg-m.	Cumulative Oper. Mom. Kg-m.	Cumulative Hydro. Mom. Kg-m.
10	20	...	...	...
20	30	...	...	...
30	40	...	...	...
40	50	...	...	...
50	60	...	...	...
60	70	...	...	...
70	80	...	...	...
80	90	...	...	...
90	100	...	...	...
100	110	...	...	...

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Conical Section: Cone: 16 4:14pm Feb 20, 2026

**Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1****Conical Section From 20 To 30 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	12.7573 mm.
Maximum Centroid Reinforcement Distance Small End	11.0651 mm.

**Reinforcement Calculations for Cone / Large Cylinder:****Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00048
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	6.136 degrees

**Required Area of Reinforcement for Large End Under External Pressure**

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.187 degrees
Large end actual half apex angle	6.136 degrees

Intermediate Value [k]:

$$= \max(Y / (S_{rl} * E_{rl}), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Large End All. Stress} * \text{Large End Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Allowable Stress of Large End Material (Ext. Temp)	133.8 N./mm <sup>2</sup>
Allowable Stress of Cone Material (Ext. Temp)	133.8 N./mm <sup>2</sup>

**Required Area of Reinforcement, Large End, External [Arl]:**

$$= (k * Q_L * R_L * \tan(\text{angle}) / (S_s * E_L)) * (1 - 1/4 * ((P * R_L - Q_L) / Q_L) * (\delta / \alpha))$$

$$= (1.0068 * 0.4426 * 868.0 * 0.107 / (134 * 1.0)) * (1 - 1/4 * ((0.1 * 868.0 - 0.443) / 0.443) * (0.187 / 6.136))$$

$$= 0.0302 \text{ cm}^2$$

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Conical Section:

Cone: 16 4:14pm Feb 20,2026

**Force per Length, Cone Large End External Pressure [QL]:**

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(868.0/2) + 0.0/(\pi(1736.0 - 3.0)) + \\
 &\quad 0.0/(\pi(868.0 - 3.0/2)(868.0 - 3.0/2)) \\
 &= 0.443 \text{ Kg/mm.}
 \end{aligned}$$

**Available Area of Reinforcement, Large End, External [Ael]:**

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1736.0 * 3.0)^{1/2} * (3.0 + 3.0/0.994) \\
 &= 2.3884 \text{ cm}^2
 \end{aligned}$$

**Summary of Reinforcement Area, Large End, External Pressure:**

Area of reinforcement required per App. 1-8(1)	0.0302	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-8(2)	2.3884	cm <sup>2</sup>
Area of reinforcement in stiffening ring	4.7000	cm <sup>2</sup>

**Intermediate Results, Large End, External Pressure:**

Area Available in Cone, Shell, and Reinforcement	46.87	cm <sup>2</sup>
Force per Unit Length on Shell / Cone Junction	1.55	Kg/mm.
Actual Buckling Stress associated with this Force	4.23	N./mm <sup>2</sup>
Material Strain associated with this stress	0.000044	

**Required Moment of Inertia, Large End, External Pressure [I's]:**

$$\begin{aligned}
 &= A * D_l^2 * A_{t1} / 10.9 \\
 &= 0.438E-04 * 1736.0 * 1736.0 * 46.87/10.9 \\
 &= 56786.08 \text{ mm}^4
 \end{aligned}$$

**Available Moment of Inertia, Large End, External Pressure:**

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di <sup>2</sup>
Shl	1.191	0.0000	0.000	16.2161	0.009	313.119
Con	1.198	-2.1334	-2.555	18.3495	0.027	403.238
Sec	4.700	25.0000	117.500	-8.7840	8.652	362.643
TOT	7.088		114.945		8.688	1079.001
Centroid of Section			16.2161	Moment of Inertia		19.48

**Summary of Large End Inertia Calculations**

Available Moment of Inertia ( Large End )	19.478	cm**4
Required Moment of Inertia ( Large End )	5.679	cm**4

**Reinforcement Calculations for Cone / Small Cylinder:****Required Area of Reinforcement for Small End under Internal Pressure**

Small end ratio of pressure to allowable stress	0.00048	
Small end max. half apex angle w/o reinforcement	4.000	degrees
Small end actual half apex angle	6.136	degrees

**Intermediate Value [k]:**

$$\begin{aligned}
 &= \max(Y / (S_r * E_{rs}), 1) \\
 &= \max(0.19226E+11 / (100.3 * 0.18458E+09), 1) \\
 &= 1.0386
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)} \\
 &= 104.2 * 0.18458E+09 \\
 &= 19225886720.0 \text{ N./mm}^2
 \end{aligned}$$

**Decay Length, Cone Small End:**

$$\begin{aligned}
 &= 1.4 * \sqrt{R_s(t_s - c_a)} \\
 &= 1.4 * \sqrt{650.0(3.0 - 0.0)} \\
 &= 61.822 \text{ mm.}
 \end{aligned}$$

**Required Area of Reinforcement, Small End, Internal [Ars]:**

$$\begin{aligned}
 &= k * Q_s * R_s / (S_s * E_1) * (1 - \delta / \alpha) * \tan(\alpha) \\
 &= 1.04 * 1.6571 * 650.0 / (104 * 1.0) *
 \end{aligned}$$

FileName : Circular Sections

Conical Section:

Cone: 16 4:14pm Feb 20,2026

$$(1 - 4.0/6.14) \cdot 0.1075$$

$$= 0.0394 \text{ cm}^2$$

**Force per Length, Cone Small End [QS]:**

$$= P(Rs/2) - Faxial/(\pi(Ds + Ts)) + Moment/(\pi(Rs + Ts/2)(Rs + Ts/2))$$

$$= 0.5(650.0/2) - 0.0/(\pi(1300.0 + 3.0)) +$$

$$0.0/(\pi(650.0 + 3.0/2)(650.0 + 3.0/2))$$

$$= 1.657 \text{ Kgf/mm.}$$

**Area of Reinforcement Available in Small End Shell [Aes]:**

$$= 0.78(Rs \cdot Ts)^{1/2} \cdot ((Ts - t) + (Tc - Tr) / \cos(\alpha))$$

$$= 0.78(650.0 \cdot 3.0)^{1/2} \cdot ((3.0 - 0.312) + (3.0 - 0.314) / 0.99)$$

$$= 1.8563 \text{ cm}^2$$

**Summary of Reinforcement Area, Small End, Internal Pressure:**

Area of reinforcement required per App. 1-5(3)	0.0394	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-5(4)	1.8563	cm <sup>2</sup>
Area of reinforcement in stiffening ring	9.7000	cm <sup>2</sup>

**Required Area of Reinforcement for Small End Under External Pressure**

Allowable Stress of Small End Material (Ext. Temp)	133.8	N./mm <sup>2</sup>
Allowable Stress of Cone Material (Ext. Temp)	133.8	N./mm <sup>2</sup>

**Intermediate Value [k]:**

$$= \max(Y / (Srs \cdot Ers), 1)$$

$$= \max(0.25813E+11 / (132.9 \cdot 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} \cdot \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 \cdot 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

**Area of Reinforcement Required in Small End Shell [Ars]:**

$$= k \cdot QS \cdot Rs \cdot \tan(\alpha) / (Ss \cdot E1)$$

$$= (1.0068 \cdot 0.333 \cdot 653.0 \cdot 0.1075) / (134 \cdot 1.0)$$

$$= 0.0172 \text{ cm}^2$$

**Force per Length, Cone Small End [QS]:**

$$= Pext(Rs/2) + Faxial/(\pi(Ds - Ts)) + Moment/(\pi(Rs - Ts/2)(Rs - Ts/2))$$

$$= 0.1(653.0/2) + 0.0/(\pi(1306.0 - 3.0)) +$$

$$0.0/(\pi(653.0 - 3.0/2)(653.0 - 3.0/2))$$

$$= 0.333 \text{ Kgf/mm.}$$

**Area of Reinforcement Available in Small End Shell [Aes]:**

$$= 0.55 \cdot (Ds \cdot ts)^{1/2} \cdot ((ts - t) + (tc - tr) / \cos(\alpha))$$

$$= 0.55 \cdot (1306.0 \cdot 3.0)^{1/2} \cdot ((3.0 - 0.573) + (3.0 - 1.893) / 0.994)$$

$$= 1.2188 \text{ cm}^2$$

**Summary of Reinforcement Area, Small End, External Pressure:**

Area of reinforcement required per App. 1-8(3)	0.0172	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-8(4)	1.2188	cm <sup>2</sup>
Area of reinforcement in stiffening ring	9.7000	cm <sup>2</sup>

**Intermediate Results, Small End, External Pressure:**

Area Available in Cone, Shell, and Reinforcement	41.37	cm <sup>2</sup>
Force per Unit Length on Shell / Cone Junction	0.88	Kgf/mm.
Actual Buckling Stress associated with this Force	2.04	N./mm <sup>2</sup>
Material Strain associated with this stress	0.000021	

**Required Moment of Inertia, Small End, External Pressure [I's]:**

$$= A \cdot Ds^2 \cdot Ats / 10.9$$

$$= 0.211E-04 \cdot 1306.0 \cdot 1306.0 \cdot 41.37 / 10.9$$

$$= 13689.88 \text{ mm}^4$$

FileName : Circular Sections

Conical Section:

Cone: 16 4:14pm Feb 20,2026

**Available Moment of Inertia, Small End, External Pressure:**

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di <sup>2</sup>
Shl	1.033	0.0000	0.000	41.3643	0.008	1767.127
Con	1.039	1.8504	1.922	39.5139	0.020	1621.850
Sec	9.700	50.0000	485.000	-8.6357	76.056	723.377
TOT	11.772		486.922		76.083	4112.354
Centroid of Section			41.3643	Moment of Inertia		117.21

**Summary of Small End Inertia Calculations**

Available Moment of Inertia ( Small End )	0.1172E+03	cm**4
Required Moment of Inertia ( Small End )	0.1369E+01	cm**4

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

**Results for Discontinuity Stresses per Bednar p. 236 2nd Edition**

Stress Type	Stress	Allowable	Location
Tensile Stress	15.45	312.47	Small Cyl. Long.
Compres. Stress	-4.60	-312.47	Small Cyl. Long.
Membrane Stress	16.38	-156.24	Small End Tang.
Tensile Stress	15.49	312.47	Cone Longitudinal
Compres. Stress	-4.56	-312.47	Cone Longitudinal
Compres Stress	16.44	-156.24	Cone Tangential
Tensile Stress	22.60	312.47	Large Cyl. Long.
Compres. Stress	-8.16	-312.47	Large Cyl. Long.
Membrane Stress	5.98	-156.24	Large End Tang.
Tensile Stress	22.64	312.47	Cone Longitudinal
Compres. Stress	-8.11	-312.47	Cone Longitudinal
Compres Stress	6.06	-156.24	Cone Tangential

Note: An asterisk (\*) denotes that this stress was not applicable for this combination of loads.

**Maximum Allowable Pressure Calculations for Cone to Shell Junction:**

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	3.584	Thickness due to internal pressure, Cone Large End
MAPnc	4.745	Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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FileName : Circular Sections

Conical Section:

Cone: 17 4:14pm Feb 20, 2026

**Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1****Conical Section From 40 To 50 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	12.7573 mm.
Maximum Centroid Reinforcement Distance Small End	11.4810 mm.

**Reinforcement Calculations for Cone / Large Cylinder:****Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00048
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	7.829 degrees

**Required Area of Reinforcement for Large End Under External Pressure**

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.187 degrees
Large end actual half apex angle	7.829 degrees

Intermediate Value [k]:

$$= \max(Y / (S_{rl} * E_{rl}), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Allowable Stress of Large End Material (Ext. Temp)	133.8 N./mm <sup>2</sup>
Allowable Stress of Cone Material (Ext. Temp)	133.8 N./mm <sup>2</sup>

**Required Area of Reinforcement, Large End, External [Arl]:**

$$= (k * Q_L * R_L * \tan(\text{angle}) / (S_s * E_L)) * (1 - 1/4 * ((P * R_L - Q_L) / Q_L) * (\delta / \alpha))$$

$$= (1.0068 * 0.4426 * 868.0 * 0.138 / (134 * 1.0)) * (1 - 1/4 * ((0.1 * 868.0 - 0.443) / 0.443) * (0.187 / 7.829))$$

$$= 0.0387 \text{ cm}^2$$

FileName : Circular Sections

Conical Section:

Cone: 17 4:14pm Feb 20,2026

**Force per Length, Cone Large End External Pressure [QL]:**

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(868.0/2) + 0.0/(\pi(1736.0 - 3.0)) + \\
 &\quad 0.0/(\pi(868.0 - 3.0/2)(868.0 - 3.0/2)) \\
 &= 0.443 \text{ Kg/mm.}
 \end{aligned}$$

**Available Area of Reinforcement, Large End, External [Ael]:**

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1736.0 * 3.0)^{1/2} * (3.0 + 3.0/0.991) \\
 &= 2.3927 \text{ cm}^2
 \end{aligned}$$

**Summary of Reinforcement Area, Large End, External Pressure:**

Area of reinforcement required per App. 1-8(1)	0.0387	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-8(2)	2.3927	cm <sup>2</sup>
Area of reinforcement in stiffening ring	5.0000	cm <sup>2</sup>

**Intermediate Results, Large End, External Pressure:**

Area Available in Cone, Shell, and Reinforcement	35.17	cm <sup>2</sup>
Force per Unit Length on Shell / Cone Junction	1.09	Kgf/mm.
Actual Buckling Stress associated with this Force	3.94	N./mm <sup>2</sup>
Material Strain associated with this stress	0.000041	

**Required Moment of Inertia, Large End, External Pressure [I's]:**

$$\begin{aligned}
 &= A * D_l^2 * A_{t1} / 10.9 \\
 &= 0.408E-04 * 1736.0 * 1736.0 * 35.17/10.9 \\
 &= 39720.57 \text{ mm}^4
 \end{aligned}$$

**Available Moment of Inertia, Large End, External Pressure:**

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di <sup>2</sup>
Shl	1.191	0.0000	0.000	17.4794	0.009	363.808
Con	1.202	-2.7288	-3.280	20.2082	0.039	490.842
Sec	5.000	26.5000	132.500	-9.0206	10.417	406.853
TOT	7.393		129.220		10.465	1261.504
Centroid of Section			17.4794	Moment of Inertia		23.08

**Summary of Large End Inertia Calculations**

Available Moment of Inertia ( Large End )	23.080	cm**4
Required Moment of Inertia ( Large End )	3.972	cm**4

**Reinforcement Calculations for Cone / Small Cylinder:****Required Area of Reinforcement for Small End under Internal Pressure**

Small end ratio of pressure to allowable stress	0.00048	
Small end max. half apex angle w/o reinforcement	4.000	degrees
Small end actual half apex angle	7.829	degrees

**Intermediate Value [k]:**

$$\begin{aligned}
 &= \max(Y / (S_r * E_{rs}), 1) \\
 &= \max(0.19226E+11 / (100.3 * 0.18458E+09), 1) \\
 &= 1.0386
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)} \\
 &= 104.2 * 0.18458E+09 \\
 &= 19225886720.0 \text{ N./mm}^2
 \end{aligned}$$

**Decay Length, Cone Small End:**

$$\begin{aligned}
 &= 1.4 * \sqrt{R_s(t_s - c_a)} \\
 &= 1.4 * \sqrt{700.0(3.0 - 0.0)} \\
 &= 64.156 \text{ mm.}
 \end{aligned}$$

**Required Area of Reinforcement, Small End, Internal [Ars]:**

$$\begin{aligned}
 &= k * Q_s * R_s / (S_s * E_1) * (1 - \delta / \alpha) * \tan(\alpha) \\
 &= 1.04 * 1.7846 * 700.0 / (104 * 1.0) *
 \end{aligned}$$

FileName : Circular Sections

Conical Section:

Cone: 17 4:14pm Feb 20,2026

$$(1 - 4.0/7.83) 0.1375$$

$$= 0.0822 \text{ cm}^2$$

**Force per Length, Cone Small End [QS]:**

$$= P(Rs/2) - Faxial/(\pi(Ds + Ts)) + Moment/(\pi(Rs + Ts/2)(Rs + Ts/2))$$

$$= 0.5(700.0/2) - 0.0/(\pi(1400.0 + 3.0)) +$$

$$0.0/(\pi(700.0 + 3.0/2)(700.0 + 3.0/2))$$

$$= 1.785 \text{ Kgf/mm.}$$

**Area of Reinforcement Available in Small End Shell [Aes]:**

$$= 0.78(Rs*Ts)^{1/2}((Ts-t)+(Tc-Tr)/\cos(\alpha))$$

$$= 0.78(700.0*3.0)^{1/2}((3.0-0.336)+(3.0-0.339)/0.99)$$

$$= 1.9122 \text{ cm}^2$$

**Summary of Reinforcement Area, Small End, Internal Pressure:**

Area of reinforcement required per App. 1-5(3)	0.0822	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-5(4)	1.9122	cm <sup>2</sup>
Area of reinforcement in stiffening ring	4.6860	cm <sup>2</sup>

**Required Area of Reinforcement for Small End Under External Pressure**

Allowable Stress of Small End Material (Ext. Temp)	133.8	N./mm <sup>2</sup>
Allowable Stress of Cone Material (Ext. Temp)	133.8	N./mm <sup>2</sup>

**Intermediate Value [k]:**

$$= \max(Y / (Srs * Ers), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

**Area of Reinforcement Required in Small End Shell [Ars]:**

$$= k * QS * Rs * \tan(\alpha) / (Ss * E1)$$

$$= (1.0068 * 0.3584 * 703.0 * 0.1375) / (134 * 1.0)$$

$$= 0.0256 \text{ cm}^2$$

**Force per Length, Cone Small End [QS]:**

$$= Pext(Rs/2) + Faxial/(\pi(Ds - Ts)) + Moment/(\pi(Rs - Ts/2)(Rs - Ts/2))$$

$$= 0.1(703.0/2) + 0.0/(\pi(1406.0 - 3.0)) +$$

$$0.0/(\pi(703.0 - 3.0/2)(703.0 - 3.0/2))$$

$$= 0.358 \text{ Kgf/mm.}$$

**Area of Reinforcement Available in Small End Shell [Aes]:**

$$= 0.55*(Ds*ts)^{1/2}[(ts-t)+(tc-tr)/\cos(\alpha)]$$

$$= 0.55*(1406.0*3.0)^{1/2}[(3.0-0.57)+(3.0-0.514)/0.991]$$

$$= 1.7647 \text{ cm}^2$$

**Summary of Reinforcement Area, Small End, External Pressure:**

Area of reinforcement required per App. 1-8(3)	0.0256	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-8(4)	1.7647	cm <sup>2</sup>
Area of reinforcement in stiffening ring	4.6860	cm <sup>2</sup>

**Intermediate Results, Small End, External Pressure:**

Area Available in Cone, Shell, and Reinforcement	24.21	cm <sup>2</sup>
Force per Unit Length on Shell / Cone Junction	0.55	Kgf/mm.
Actual Buckling Stress associated with this Force	2.35	N./mm <sup>2</sup>
Material Strain associated with this stress	0.000024	

**Required Moment of Inertia, Small End, External Pressure [I's]:**

$$= A * Ds^2 * Ats / 10.9$$

$$= 0.244E-04 * 1406.0 * 1406.0 * 24.21/10.9$$

$$= 10712.14 \text{ mm}^4$$

FileName : Circular Sections

Conical Section:

Cone: 17 4:14pm Feb 20,2026

**Available Moment of Inertia, Small End, External Pressure:**

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di <sup>2</sup>
Shl	1.072	0.0000	0.000	17.4694	0.008	327.033
Con	1.082	2.4558	2.656	15.0136	0.030	243.822
Sec	4.686	24.9300	116.822	-7.4606	8.575	260.827
TOT	6.839		119.478		8.613	831.682
Centroid of Section			17.4694	Moment of Inertia		16.93

**Summary of Small End Inertia Calculations**

Available Moment of Inertia ( Small End )	0.1693E+02	cm**4
Required Moment of Inertia ( Small End )	0.1071E+01	cm**4

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

**Results for Discontinuity Stresses per Bednar p. 236 2nd Edition**

Stress Type	Stress	Allowable	Location
Tensile Stress	20.16	312.47	Small Cyl. Long.
Compres. Stress	-8.47	-312.47	Small Cyl. Long.
Membrane Stress	19.57	-156.24	Small End Tang.
Tensile Stress	20.22	312.47	Cone Longitudinal
Compres. Stress	-8.41	-312.47	Cone Longitudinal
Compres Stress	19.68	-156.24	Cone Tangential
Tensile Stress	26.87	312.47	Large Cyl. Long.
Compres. Stress	-12.43	-312.47	Large Cyl. Long.
Membrane Stress	3.63	-156.24	Large End Tang.
Tensile Stress	26.94	312.47	Cone Longitudinal
Compres. Stress	-12.36	-312.47	Cone Longitudinal
Compres Stress	3.76	-156.24	Cone Tangential

Note: An asterisk (\*) denotes that this stress was not applicable for this combination of loads.

**Maximum Allowable Pressure Calculations for Cone to Shell Junction:**

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	3.571	Thickness due to internal pressure, Cone Large End
MAPnc	4.728	Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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FileName : Circular Sections -----

Conical Section: Cone: 18 4:14pm Feb 20, 2026

**Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1****Conical Section From 60 To 70 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	12.7573 mm.
Maximum Centroid Reinforcement Distance Small End	11.4810 mm.

**Reinforcement Calculations for Cone / Large Cylinder:****Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00048
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	4.716 degrees

**Required Area of Reinforcement for Large End Under External Pressure**

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.187 degrees
Large end actual half apex angle	4.716 degrees

Intermediate Value [k]:

$$= \max(Y / (S_{rl} * E_{rl}), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Allowable Stress of Large End Material (Ext. Temp)	133.8 N./mm <sup>2</sup>
Allowable Stress of Cone Material (Ext. Temp)	133.8 N./mm <sup>2</sup>

**Required Area of Reinforcement, Large End, External [Arl]:**

$$= (k * Q_L * R_L * \tan(\text{angle}) / (S_s * E_L)) * (1 - 1/4 * ((P * R_L - Q_L) / Q_L) * (\Delta / \alpha))$$

$$= (1.0068 * 0.4426 * 868.0 * 0.082 / (134 * 1.0)) * (1 - 1/4 * ((0.1 * 868.0 - 0.443) / 0.443) * (0.187 / 4.716))$$

$$= 0.0232 \text{ cm}^2$$

FileName : Circular Sections

Conical Section:

Cone: 18 4:14pm Feb 20,2026

**Force per Length, Cone Large End External Pressure [QL]:**

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(868.0/2) + 0.0/(\pi(1736.0 - 3.0)) + \\
 &\quad 0.0/(\pi(868.0 - 3.0/2)(868.0 - 3.0/2)) \\
 &= 0.443 \text{ Kg/mm.}
 \end{aligned}$$

**Available Area of Reinforcement, Large End, External [Ael]:**

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1736.0 * 3.0)^{1/2} * (3.0 + 3.0/0.997) \\
 &= 2.3855 \text{ cm}^2
 \end{aligned}$$

**Summary of Reinforcement Area, Large End, External Pressure:**

Area of reinforcement required per App. 1-8(1)	0.0232	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-8(2)	2.3855	cm <sup>2</sup>
Area of reinforcement in stiffening ring	4.7000	cm <sup>2</sup>

**Intermediate Results, Large End, External Pressure:**

Area Available in Cone, Shell, and Reinforcement	45.86	cm <sup>2</sup>
Force per Unit Length on Shell / Cone Junction	1.57	Kgf/mm.
Actual Buckling Stress associated with this Force	4.37	N./mm <sup>2</sup>
Material Strain associated with this stress	0.000045	

**Required Moment of Inertia, Large End, External Pressure [I's]:**

$$\begin{aligned}
 &= A * D_l^2 * A_{t1} / 10.9 \\
 &= 0.453E-04 * 1736.0 * 1736.0 * 45.86/10.9 \\
 &= 57448.63 \text{ mm}^4
 \end{aligned}$$

**Available Moment of Inertia, Large End, External Pressure:**

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di <sup>2</sup>
Shl	1.191	0.0000	0.000	16.3070	0.009	316.641
Con	1.195	-1.6373	-1.956	17.9443	0.020	384.720
Sec	4.700	25.0000	117.500	-8.6930	8.652	355.172
TOT	7.086		115.544		8.681	1056.534
Centroid of Section			16.3070	Moment of Inertia		19.25

**Summary of Large End Inertia Calculations**

Available Moment of Inertia ( Large End )	19.246	cm**4
Required Moment of Inertia ( Large End )	5.745	cm**4

**Reinforcement Calculations for Cone / Small Cylinder:****Required Area of Reinforcement for Small End under Internal Pressure**

Small end ratio of pressure to allowable stress	0.00048	
Small end max. half apex angle w/o reinforcement	4.000	degrees
Small end actual half apex angle	4.716	degrees

**Intermediate Value [k]:**

$$\begin{aligned}
 &= \max(Y / (S_r * E_{rs}), 1) \\
 &= \max(0.19226E+11 / (100.3 * 0.18458E+09), 1) \\
 &= 1.0386
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)} \\
 &= 104.2 * 0.18458E+09 \\
 &= 19225886720.0 \text{ N./mm}^2
 \end{aligned}$$

**Decay Length, Cone Small End:**

$$\begin{aligned}
 &= 1.4 * \sqrt{R_s(t_s - c_a)} \\
 &= 1.4 * \sqrt{700.0(3.0 - 0.0)} \\
 &= 64.156 \text{ mm.}
 \end{aligned}$$

**Required Area of Reinforcement, Small End, Internal [Ars]:**

$$\begin{aligned}
 &= k * Q_s * R_s / (S_s * E_1) * (1 - \delta / \alpha) * \tan(\alpha) \\
 &= 1.04 * 1.7846 * 700.0 / (104 * 1.0) *
 \end{aligned}$$

FileName : Circular Sections

Conical Section:

Cone: 18 4:14pm Feb 20,2026

$$(1 - 4.0/4.72) 0.0825$$

$$= 0.0153 \text{ cm}^2$$

**Force per Length, Cone Small End [QS]:**

$$= P(Rs/2) - Faxial/(\pi(Ds + Ts)) + Moment/(\pi(Rs + Ts/2)(Rs + Ts/2))$$

$$= 0.5(700.0/2) - 0.0/(\pi(1400.0 + 3.0)) +$$

$$0.0/(\pi(700.0 + 3.0/2)(700.0 + 3.0/2))$$

$$= 1.785 \text{ Kgf/mm.}$$

**Area of Reinforcement Available in Small End Shell [Aes]:**

$$= 0.78(Rs*Ts)^{1/2} * ((Ts-t) + (Tc-Tr)/\cos(\alpha))$$

$$= 0.78(700.0*3.0)^{1/2} * ((3.0 - 0.336) + (3.0 - 0.337)/1.)$$

$$= 1.9072 \text{ cm}^2$$

**Summary of Reinforcement Area, Small End, Internal Pressure:**

Area of reinforcement required per App. 1-5(3)	0.0153	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-5(4)	1.9072	cm <sup>2</sup>
Area of reinforcement in stiffening ring	4.7000	cm <sup>2</sup>

**Required Area of Reinforcement for Small End Under External Pressure**

Allowable Stress of Small End Material (Ext. Temp)	133.8	N./mm <sup>2</sup>
Allowable Stress of Cone Material (Ext. Temp)	133.8	N./mm <sup>2</sup>

**Intermediate Value [k]:**

$$= \max(Y / (Srs * Ers), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

**Area of Reinforcement Required in Small End Shell [Ars]:**

$$= k * QS * Rs * \tan(\alpha) / (Ss * E1)$$

$$= (1.0068 * 0.3584 * 703.0 * 0.0825) / (134 * 1.0)$$

$$= 0.0153 \text{ cm}^2$$

**Force per Length, Cone Small End [QS]:**

$$= Pext(Rs/2) + Faxial/(\pi(Ds - Ts)) + Moment/(\pi(Rs - Ts/2)(Rs - Ts/2))$$

$$= 0.1(703.0/2) + 0.0/(\pi(1406.0 - 3.0)) +$$

$$0.0/(\pi(703.0 - 3.0/2)(703.0 - 3.0/2))$$

$$= 0.358 \text{ Kgf/mm.}$$

**Area of Reinforcement Available in Small End Shell [Aes]:**

$$= 0.55 * (Ds*ts)^{1/2} * ((ts-t) + (tc-tr)/\cos(\alpha))$$

$$= 0.55 * (1406.0 * 3.0)^{1/2} * ((3.0 - 1.89) + (3.0 - 2.0)/0.997)$$

$$= 0.7552 \text{ cm}^2$$

**Summary of Reinforcement Area, Small End, External Pressure:**

Area of reinforcement required per App. 1-8(3)	0.0153	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-8(4)	0.7552	cm <sup>2</sup>
Area of reinforcement in stiffening ring	4.7000	cm <sup>2</sup>

**Intermediate Results, Small End, External Pressure:**

Area Available in Cone, Shell, and Reinforcement	58.61	cm <sup>2</sup>
Force per Unit Length on Shell / Cone Junction	1.60	Kgf/mm.
Actual Buckling Stress associated with this Force	2.82	N./mm <sup>2</sup>
Material Strain associated with this stress	0.000029	

**Required Moment of Inertia, Small End, External Pressure [I's]:**

$$= A * Ds^2 * Ats / 10.9$$

$$= 0.292E-04 * 1406.0 * 1406.0 * 58.61 / 10.9$$

$$= 31077.31 \text{ mm}^4$$

FileName : Circular Sections

Conical Section:

Cone: 18 4:14pm Feb 20,2026

**Available Moment of Inertia, Small End, External Pressure:**

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di <sup>2</sup>
Shl	1.072	0.0000	0.000	17.3925	0.008	324.163
Con	1.075	1.4735	1.584	15.9191	0.016	272.487
Sec	4.700	25.0000	117.500	-7.6075	8.652	272.005
TOT	6.847		119.084		8.676	868.655
Centroid of Section			17.3925	Moment of Inertia		17.36

**Summary of Small End Inertia Calculations**

Available Moment of Inertia ( Small End )	0.1736E+02	cm**4
Required Moment of Inertia ( Small End )	0.3108E+01	cm**4

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

**Results for Discontinuity Stresses per Bednar p. 236 2nd Edition**

Stress Type	Stress	Allowable	Location
Tensile Stress	14.45	312.47	Small Cyl. Long.
Compres. Stress	-2.76	-312.47	Small Cyl. Long.
Membrane Stress	16.43	-156.24	Small End Tang.
Tensile Stress	14.47	312.47	Cone Longitudinal
Compres. Stress	-2.74	-312.47	Cone Longitudinal
Compres Stress	16.47	-156.24	Cone Tangential
Tensile Stress	19.03	312.47	Large Cyl. Long.
Compres. Stress	-4.59	-312.47	Large Cyl. Long.
Membrane Stress	7.94	-156.24	Large End Tang.
Tensile Stress	19.05	312.47	Cone Longitudinal
Compres. Stress	-4.56	-312.47	Cone Longitudinal
Compres Stress	7.99	-156.24	Cone Tangential

Note: An asterisk (\*) denotes that this stress was not applicable for this combination of loads.

**Maximum Allowable Pressure Calculations for Cone to Shell Junction:**

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	3.592	Thickness due to internal pressure, Cone Large End
MAPnc	4.756	Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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FileName : Circular Sections -----

Conical Section: Cone: 19 4:14pm Feb 20, 2026

**Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1****Conical Section From 80 To 90 SA-240 316**

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	12.7573 mm.
Maximum Centroid Reinforcement Distance Small End	11.4810 mm.

**Reinforcement Calculations for Cone / Large Cylinder:****Required Area of Reinforcement for Large End Under Internal Pressure**

Large end ratio of pressure to allowable stress	0.00048
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	7.829 degrees

**Required Area of Reinforcement for Large End Under External Pressure**

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.187 degrees
Large end actual half apex angle	7.829 degrees

Intermediate Value [k]:

$$= \max(Y / (S_{rl} * E_{rl}), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

Allowable Stress of Large End Material (Ext. Temp)	133.8 N./mm <sup>2</sup>
Allowable Stress of Cone Material (Ext. Temp)	133.8 N./mm <sup>2</sup>

**Required Area of Reinforcement, Large End, External [Arl]:**

$$= (k * Q_L * R_L * \tan(\text{angle}) / (S_s * E_L)) * (1 - 1/4 * ((P * R_L - Q_L) / Q_L) * (\delta / \alpha))$$

$$= (1.0068 * 0.4426 * 868.0 * 0.138 / (134 * 1.0)) * (1 - 1/4 * ((0.1 * 868.0 - 0.443) / 0.443) * (0.187 / 7.829))$$

$$= 0.0387 \text{ cm}^2$$

FileName : Circular Sections

Conical Section:

Cone: 19 4:14pm Feb 20,2026

**Force per Length, Cone Large End External Pressure [QL]:**

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(868.0/2) + 0.0/(\pi(1736.0 - 3.0)) + \\
 &\quad 0.0/(\pi(868.0 - 3.0/2)(868.0 - 3.0/2)) \\
 &= 0.443 \text{ Kg/mm.}
 \end{aligned}$$

**Available Area of Reinforcement, Large End, External [Ael]:**

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1736.0 * 3.0)^{1/2} * (3.0 + 3.0/0.991) \\
 &= 2.3927 \text{ cm}^2
 \end{aligned}$$

**Summary of Reinforcement Area, Large End, External Pressure:**

Area of reinforcement required per App. 1-8(1)	0.0387	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-8(2)	2.3927	cm <sup>2</sup>
Area of reinforcement in stiffening ring	4.6835	cm <sup>2</sup>

**Intermediate Results, Large End, External Pressure:**

Calculate Cylinder Decay Length:

$$\begin{aligned}
 &= 2 * \sqrt{R_l * t_s} \\
 &= 2 * \sqrt{868.0 * 3.0} \\
 &= 102.0588 \text{ mm.}
 \end{aligned}$$

Note: LL is zero since large cylinder length is less than the decay length

Area Available in Cone, Shell, and Reinforcement	22.85	cm <sup>2</sup>
Force per Unit Length on Shell / Cone Junction	0.68	Kgf/mm.
Actual Buckling Stress associated with this Force	3.78	N./mm <sup>2</sup>
Material Strain associated with this stress	0.000039	

**Required Moment of Inertia, Large End, External Pressure [I's]:**

$$\begin{aligned}
 &= A * D_l^2 * A_{tl} / 10.9 \\
 &= 0.392E-04 * 1736.0 * 1736.0 * 22.85/10.9 \\
 &= 24792.07 \text{ mm.}^4
 \end{aligned}$$

**Available Moment of Inertia, Large End, External Pressure:**

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di <sup>2</sup>
Shl	1.191	0.0000	0.000	16.0286	0.009	305.921
Con	1.202	-2.7288	-3.280	18.7574	0.039	422.893
Sec	4.684	24.9175	116.701	-8.8889	8.561	370.059
TOT	7.076		113.421		8.609	1098.873
Centroid of Section			16.0286	Moment of Inertia		19.60

**Summary of Large End Inertia Calculations**

Available Moment of Inertia ( Large End )	19.598	cm**4
Required Moment of Inertia ( Large End )	2.479	cm**4

**Reinforcement Calculations for Cone / Small Cylinder:****Required Area of Reinforcement for Small End under Internal Pressure**

Small end ratio of pressure to allowable stress	0.00048	
Small end max. half apex angle w/o reinforcement	4.000	degrees
Small end actual half apex angle	7.829	degrees

Intermediate Value [k]:

$$\begin{aligned}
 &= \max(Y / (S_r * E_{rs}), 1) \\
 &= \max(0.19226E+11 / (100.3 * 0.18458E+09), 1) \\
 &= 1.0386
 \end{aligned}$$

where [Y] is:

$$\begin{aligned}
 &= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Int. temp.)} \\
 &= 104.2 * 0.18458E+09 \\
 &= 19225886720.0 \text{ N./mm}^2
 \end{aligned}$$

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FileName : Circular Sections -----

Conical Section: Cone: 19 4:14pm Feb 20,2026

Decay Length, Cone Small End:

$$= 1.4 * \sqrt{Rs(t_s - c_a)}$$

$$= 1.4 * \sqrt{700.0(3.0 - 0.0)}$$

$$= 64.156 \text{ mm.}$$

**Required Area of Reinforcement, Small End, Internal [Ars]:**

$$= k * Q_s * R_s / (S_s * E_1) * (1 - \delta / \alpha) * \tan(\alpha)$$

$$= 1.04 * 1.7846 * 700.0 / (104 * 1.0) * (1 - 4.0/7.83) * 0.1375$$

$$= 0.0822 \text{ cm}^2$$

**Force per Length, Cone Small End [QS]:**

$$= P(R_s/2) - F_{axial} / (\pi(D_s + T_s)) + \text{Moment} / (\pi(R_s + T_s/2)(R_s + T_s/2))$$

$$= 0.5(700.0/2) - 0.0 / (\pi(1400.0 + 3.0)) + 0.0 / (\pi(700.0 + 3.0/2)(700.0 + 3.0/2))$$

$$= 1.785 \text{ Kgf/mm.}$$

**Area of Reinforcement Available in Small End Shell [Aes]:**

$$= 0.78(R_s * T_s)^{1/2} * ((T_s - t) + (T_c - T_r) / \cos(\alpha))$$

$$= 0.78(700.0 * 3.0)^{1/2} * ((3.0 - 0.336) + (3.0 - 0.339) / 0.99)$$

$$= 1.9122 \text{ cm}^2$$

**Summary of Reinforcement Area, Small End, Internal Pressure:**

Area of reinforcement required per App. 1-5(3)	0.0822	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-5(4)	1.9122	cm <sup>2</sup>
Area of reinforcement in stiffening ring	4.6830	cm <sup>2</sup>

**Required Area of Reinforcement for Small End Under External Pressure**

Allowable Stress of Small End Material (Ext. Temp)	133.8	N./mm <sup>2</sup>
Allowable Stress of Cone Material (Ext. Temp)	133.8	N./mm <sup>2</sup>

Intermediate Value [k]:

$$= \max(Y / (S_{rs} * E_{rs}), 1)$$

$$= \max(0.25813E+11 / (132.9 * 0.19291E+09), 1)$$

$$= 1.0068$$

where [Y] is:

$$= \text{Cone Allowable Stress} * \text{Cone Elastic Modulus (Ext. temp.)}$$

$$= 133.8 * 0.19291E+09$$

$$= 25812725760.0 \text{ N./mm}^2$$

**Area of Reinforcement Required in Small End Shell [Ars]:**

$$= k * Q_s * R_s * \tan(\alpha) / (S_s * E_1)$$

$$= (1.0068 * 0.3584 * 703.0 * 0.1375) / (134 * 1.0)$$

$$= 0.0256 \text{ cm}^2$$

**Force per Length, Cone Small End [QS]:**

$$= P_{ext}(R_s/2) + F_{axial} / (\pi(D_s - T_s)) + \text{Moment} / (\pi(R_s - T_s/2)(R_s - T_s/2))$$

$$= 0.1(703.0/2) + 0.0 / (\pi(1406.0 - 3.0)) + 0.0 / (\pi(703.0 - 3.0/2)(703.0 - 3.0/2))$$

$$= 0.358 \text{ Kgf/mm.}$$

**Area of Reinforcement Available in Small End Shell [Aes]:**

$$= 0.55 * (D_s * t_s)^{1/2} * [(t_s - t) + (t_c - t_r) / \cos(\alpha)]$$

$$= 0.55 * (1406.0 * 3.0)^{1/2} * [(3.0 - 1.043) + (3.0 - 0.514) / 0.991]$$

$$= 1.5954 \text{ cm}^2$$

**Summary of Reinforcement Area, Small End, External Pressure:**

Area of reinforcement required per App. 1-8(3)	0.0256	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-8(4)	1.5954	cm <sup>2</sup>
Area of reinforcement in stiffening ring	4.6830	cm <sup>2</sup>

**Intermediate Results, Small End, External Pressure:**

Area Available in Cone, Shell, and Reinforcement	28.48	cm <sup>2</sup>
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FileName : Circular Sections

Conical Section:

Cone: 19 4:14pm Feb 20,2026

Force per Unit Length on Shell / Cone Junction 0.70 Kgf/mm.  
 Actual Buckling Stress associated with this Force 2.53 N./mm<sup>2</sup>  
 Material Strain associated with this stress 0.000026

**Required Moment of Inertia, Small End, External Pressure [I's]:**

$$= A * D_s^2 * A_{ts} / 10.9$$

$$= 0.262E-04 * 1406.0 * 1406.0 * 28.48/10.9$$

$$= 13537.53 \text{ mm}^4$$

**Available Moment of Inertia, Small End, External Pressure:**

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di <sup>2</sup>
Shl	1.072	0.0000	0.000	17.4558	0.008	326.525
Con	1.082	2.4558	2.656	15.0000	0.030	243.382
Sec	4.683	24.9150	116.677	-7.4592	8.558	260.558
TOT	6.836		119.333		8.596	830.465
Centroid of Section			17.4558	Moment of Inertia		16.90

**Summary of Small End Inertia Calculations**

Available Moment of Inertia ( Small End )	0.1690E+02	cm**4
Required Moment of Inertia ( Small End )	0.1354E+01	cm**4

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

**Results for Discontinuity Stresses per Bednar p. 236 2nd Edition**

Stress Type	Stress	Allowable	Location
Tensile Stress	20.16	312.47	Small Cyl. Long.
Compres. Stress	-8.47	-312.47	Small Cyl. Long.
Membrane Stress	19.57	-156.24	Small End Tang.
Tensile Stress	20.22	312.47	Cone Longitudinal
Compres. Stress	-8.41	-312.47	Cone Longitudinal
Compres Stress	19.68	-156.24	Cone Tangential
Tensile Stress	26.87	312.47	Large Cyl. Long.
Compres. Stress	-12.43	-312.47	Large Cyl. Long.
Membrane Stress	3.63	-156.24	Large End Tang.
Tensile Stress	26.94	312.47	Cone Longitudinal
Compres. Stress	-12.36	-312.47	Cone Longitudinal
Compres Stress	3.76	-156.24	Cone Tangential

Note: An asterisk (\*) denotes that this stress was not applicable for this combination of loads.

**Maximum Allowable Pressure Calculations for Cone to Shell Junction:**

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	3.571	Thickness due to internal pressure, Cone Large End
MAPnc	4.728	Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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FileName : Circular Sections -----

Conical Section: Cone: 20 4:14pm Feb 20, 2026

### Conical Reinforcement Calculations, ASME VIII Div. 1, App. 1

#### Conical Section From 100 To 110 SA-240 316

Elastic Modulus Data from ASME Section II Part D at 50 °C

Elastic Modulus of Cone Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large Cylinder Material	0.193E+09 KPa. at 50 °C
Elastic Modulus of Large End Reinforcement	0.193E+09 KPa. at 50 °C
Elastic Modulus of Small End Reinforcement	0.193E+09 KPa. at 50 °C

Elastic Modulus Data from ASME Section II Part D at 170 °C

Elastic Modulus of Cone Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large Cylinder Material	0.185E+09 KPa. at 170 °C
Elastic Modulus of Large End Reinforcement	0.185E+09 KPa. at 170 °C
Elastic Modulus of Small End Reinforcement	0.185E+09 KPa. at 170 °C

Axial Force on Small End of Cone	0.00 Kgf
Axial Force on Large End of Cone	0.00 Kgf
Moment on Small End of Cone	0.00 Kg-m.
Moment on Large End of Cone	0.00 Kg-m.

Note: Since there are no supports, the forces and moments could not be resolved to the large and small ends of the cone.

Note: Both ends of the Cone are Lines of Support

Maximum Centroid Reinforcement Distance Large End	11.4810 mm.
Maximum Centroid Reinforcement Distance Small End	9.4670 mm.

#### Reinforcement Calculations for Cone / Large Cylinder:

##### Required Area of Reinforcement for Large End Under Internal Pressure

Large end ratio of pressure to allowable stress	0.00048
Large end max. half apex angle w/o reinforcement	11.000 degrees
Large end actual half apex angle	6.419 degrees

##### Required Area of Reinforcement for Large End Under External Pressure

Large end ratio of pressure to allowable stress	0.00007
Large end max. half apex angle w/o reinforcement	0.187 degrees
Large end actual half apex angle	6.419 degrees

Intermediate Value [k]:

= max( Y / ( Srl \* Erl ), 1 )  
 = max( 0.25813E+11 / ( 132.9 \* 0.19291E+09 ), 1 )  
 = 1.0068

where [Y] is:

= Cone Allowable Stress \* Cone Elastic Modulus (Ext. temp.)  
 = 133.8 \* 0.19291E+09  
 = 25812725760.0 N./mm<sup>2</sup>

Allowable Stress of Large End Material (Ext. Temp)	133.8 N./mm <sup>2</sup>
Allowable Stress of Cone Material (Ext. Temp)	133.8 N./mm <sup>2</sup>

##### Required Area of Reinforcement, Large End, External [Arl]:

= (k\*QL\*Rl\*tan(angle)/(Ss\*E1))\*(1-1/4((P\*Rl-QL)/QL))\*(delta/alpha)  
 = (1.0068\*0.3584\*703.0\*0.112/(134\*1.0))\*  
 (1-1/4((0.1\*703.0-0.358)/0.358))\*(0.187/6.419)  
 = 0.0208 cm<sup>2</sup>

FileName : Circular Sections

Conical Section:

Cone: 20 4:14pm Feb 20,2026

**Force per Length, Cone Large End External Pressure [QL]:**

$$\begin{aligned}
 &= P_{ext}(R_l/2) + F_{axial}/(\pi(D_l - T_l)) + \text{Moment}/(\pi(R_l - T_l/2)(R_l - T_l/2)) \\
 &= 0.1(703.0/2) + 0.0/(\pi(1406.0 - 3.0)) + \\
 &\quad 0.0/(\pi(703.0 - 3.0/2)(703.0 - 3.0/2)) \\
 &= 0.358 \text{ Kg/mm.}
 \end{aligned}$$

**Available Area of Reinforcement, Large End, External [Ael]:**

$$\begin{aligned}
 &= 0.55 * (D_l * t_s)^{1/2} * (t_s + t_c / \cos(\alpha)) \\
 &= 0.55 * (1406.0 * 3.0)^{1/2} * (3.0 + 3.0/0.994) \\
 &= 2.1500 \text{ cm}^2
 \end{aligned}$$

**Summary of Reinforcement Area, Large End, External Pressure:**

Area of reinforcement required per App. 1-8(1)	0.0208	cm <sup>2</sup>
Area of reinforcement in shell per App. 1-8(2)	2.1500	cm <sup>2</sup>
Area of reinforcement in stiffening ring	5.0420	cm <sup>2</sup>

**Intermediate Results, Large End, External Pressure:**

Area Available in Cone, Shell, and Reinforcement	61.56	cm <sup>2</sup>
Force per Unit Length on Shell / Cone Junction	2.00	Kg/mm.
Actual Buckling Stress associated with this Force	3.35	N./mm <sup>2</sup>
Material Strain associated with this stress	0.000035	

**Required Moment of Inertia, Large End, External Pressure [I's]:**

$$\begin{aligned}
 &= A * D_l^2 * A_{t1} / 10.9 \\
 &= 0.348E-04 * 1406.0 * 1406.0 * 61.56/10.9 \\
 &= 38820.20 \text{ mm}^4
 \end{aligned}$$

**Available Moment of Inertia, Large End, External Pressure:**

	Area	Centroid	Ar*Ce	Dist	I	Ar*Di <sup>2</sup>
Shl	1.072	0.0000	0.000	18.4240	0.008	363.751
Con	1.078	-2.0093	-2.167	20.4333	0.023	450.240
Sec	5.042	26.7100	134.672	-8.2860	10.681	346.172
TOT	7.192		132.505		10.712	1160.163
Centroid of Section			18.4240	Moment of Inertia		22.31

**Summary of Large End Inertia Calculations**

Available Moment of Inertia ( Large End )	22.314	cm <sup>4</sup>
Required Moment of Inertia ( Large End )	3.882	cm <sup>4</sup>

Note: The following calculations are only required per 1-5(g)(1) and do include external loads due to wind or seismic. These discontinuity stresses are computed at the shell/cone junction and do not include effects of local stiffening from a junction ring.

**Results for Discontinuity Stresses per Bednar p. 236 2nd Edition**

Stress Type	Stress	Allowable	Location
Tensile Stress	17.57	312.47	Large Cyl. Long.
Compres. Stress	-5.87	-312.47	Large Cyl. Long.
Membrane Stress	5.24	-156.24	Large End Tang.
Tensile Stress	17.60	312.47	Cone Longitudinal
Compres. Stress	-5.84	-312.47	Cone Longitudinal
Compres Stress	5.31	-156.24	Cone Tangential

Note: An asterisk (\*) denotes that this stress was not applicable for this combination of loads.

**Maximum Allowable Pressure Calculations for Cone to Shell Junction:**

Pressure Case	Pressure bars	Reason for Failure at this Pressure
MAWP	4.424	Thickness due to internal pressure, Cone Large End

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Conical Section: Cone: 20 4:14pm Feb 20,2026

MAPnc 5.857 Thickness due to internal pressure, Cone Large End

These pressures were determined by iteration.

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Center of Gravity Calculation: Step: 21 4:14pm Feb 20,2026

**Shop/Field Installation Options :**

Note : The CG is computed from the first Element From Node

Center of Gravity of Stiffening Rings	14631.545 mm.
Center of Gravity of Bare Shell New and Cold	13151.735 mm.
Center of Gravity of Bare Shell Corroded	13151.735 mm.
Vessel CG in the Operating Condition	13342.544 mm.
Vessel CG in the Fabricated (Shop/Empty) Condition	13342.544 mm.

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MDMT Summary:

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**Minimum Design Metal Temperature Results Summary :**

Description	Curve	Basic MDMT °C	Reduced MDMT °C	UG-20 (f) MDMT °C	Thickness ratio	Gov Thk mm.	E*	PWHT reqd
Notes								
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						
[16]		-196						

Warmest MDMT: -196

Required Minimum Design Metal Temperature -28.9 °C

Warmest Computed Minimum Design Metal Temperature -196.0 °C

**Notes:**

- [ ! ] - This was an impact tested material.
- [ 1 ] - Governing Nozzle Weld.
- [ 4 ] - ANSI Flange MDMT Calcs; Thickness ratio per UCS-66(b)(1)(-c).
- [ 5 ] - ANSI Flange MDMT Calcs; Thickness ratio per UCS-66(b)(1)(-b).
- [ 6 ] - MDMT Calculations at the Shell/Head Joint.
- [ 7 ] - MDMT Calculations for the Straight Flange.
- [ 8 ] - Cylinder/Cone/Flange Junction MDMT.
- [ 9 ] - Calculations in the Spherical Portion of the Head.
- [10] - Calculations in the Knuckle Portion of the Head.
- [11] - Calculated (Body Flange) Flange MDMT.
- [12] - Calculated Flat Head MDMT per UCS-66.3
- [13] - Tubesheet MDMT, shell side, if applicable
- [14] - Tubesheet MDMT, tube side, if applicable
- [15] - Nozzle Material
- [16] - Shell or Head Material
- [17] - Impact Testing required
- [18] - Impact Testing not required, see UCS-66(b)(3)
- [19] - Select a valid hydrotest type to get the UG-20(f) exemption
- [20] - Cylinder/Cone Junction MDMT based on Longitudinal Stress considerations
- [21] - Bolting Material

UG-84(b)(2) was not considered.

UCS-66(g) was not considered.

UCS-66(i) was not considered.

**Notes:**

- Impact test temps were not entered in and not considered in the analysis.
- UCS-66(i) applies to impact tested materials not by specification and
- UCS-66(g) applies to materials impact tested per UG-84.1 General Note (c).
- The Basic MDMT includes the (30F) PWHT credit if applicable.

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Vessel Design Summary:

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**ASME Code, Section VIII Division 1, 2017**

Diameter Spec : 1300.000 x 1730.000 x 1400.000 x 1730.000 x 1400.000 mm. ID  
 Vessel Design Length, Tangent to Tangent 26052.00 mm.

Distance of Bottom Tangent above Grade 8052.00 mm.  
 Distance of Base above Grade 0.00 mm.  
 Specified Datum Line Distance 0.00 mm.

Internal Design Temperature 170 °C  
 Internal Design Pressure 0.500 bars

External Design Temperature 50 °C  
 External Design Pressure 0.100 bars

Maximum Allowable Working Pressure 3.571 bars  
 External Max. Allowable Working Pressure 0.137 bars  
 Hydrostatic Test Pressure 0.000 bars

Required Minimum Design Metal Temperature -28.9 °C  
 Warmest Computed Minimum Design Metal Temperature -196.0 °C

Wind Design Code ASCE-93  
 Earthquake Design Code UBC-94

**Materials of Construction:**

Component Type	Material	Class	Thickness	UNS #	Normal ized	Impact Tested
Shell	SA-240 316	...	...	S31600	No	No
Cone	SA-240 316	...	...	S31600	No	No
Rings	SA-240 316	...	...	S31600	No	No
Rings	SA-240 304	...	...	S30400	No	No

Normalized is determined based on the UCS-66 material curve selection and Figure UCS-66.

Impact Tested is based on material selection and material data properties.

**Element Pressures and MAWP (bars & mm.):**

Element Description or Type	Design Pressure + Stat. head	Ext. Press.	Element M.A.W.P	Corrosion Allowance	Str. Flg. Gov.	In Creep Range
Cylinder	0.500	0.10	4.794	0.0000	N/A	No
Conical	0.500	0.10	3.584	0.0000	N/A	No
Cylinder	0.500	0.10	3.605	0.0000	N/A	No
Conical	0.500	0.10	3.571	0.0000	N/A	No
Cylinder	0.500	0.10	4.452	0.0000	N/A	No
Conical	0.500	0.10	3.593	0.0000	N/A	No
Cylinder	0.500	0.10	3.605	0.0000	N/A	No
Conical	0.500	0.10	3.571	0.0000	N/A	No
Cylinder	0.500	0.10	4.452	0.0000	N/A	No
Conical	0.500	0.10	4.424	0.0000	N/A	No

**Stiffener Ring Specifications:**

Elevation mm.	Selected Type	User Description
3958.00	Bar 47.0 x 10.	Ring:[1 of 1]
5946.00	Bar 47.0 x 10.	Ring:[2 of 2]
7939.00	Bar 47.0 x 10.	Ring:[3 of 3]
7952.00	Bar 47.0 x 10.	Ring:[4 of 4]
8052.00	Bar 97.0 x 10.	Ring 16
10052.00	Bar 47.0 x 10.	Ring5
12051.00	Bar 46.9 x 10.	Ring:17

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10852.00	Bar	50.0 x 10.	Ring 21
12142.00	Bar	47.0 x 10.	Ring 6
12155.00	Bar	47.0 x 10.	Ring 7
13565.00	Bar	47.0 x 10.	Ring 8
14965.00	Bar	47.0 x 10.	Ring 9
16552.00	Bar	47.0 x 10.	Ring 18
18551.00	Bar	47.0 x 10.	Ring 23
19289.00	Bar	47.0 x 10.	Ring 10
19302.00	Bar	47.0 x 10.	Ring 11
19352.00	Bar	46.8 x 10.	Ring 24
20551.00	Bar	46.8 x 10.	Ring 26
20927.00	Bar	47.0 x 10.	Ring 12
22297.00	Bar	47.0 x 10.	Ring 13]
26051.00	Bar	50.4 x 10.	Ring 22
24052.00	Bar	50.4 x 10.	Ring 28

**Element Types and Properties:**

Element Type	"To" Elev mm.	Element Length mm.	Nominal Thickness mm.	Finished Thickness mm.	Reqd Thk Internal mm.	Reqd Thk External mm.	Long Eff	Circ Eff
Cylinder	8052.0	8052.0	3.0	3.0	1.5	...	1.00	1.00
Conical	10052.0	2000.0	3.0	3.0	1.5	...	1.00	1.00
Cylinder	10852.0	800.0	3.0	3.0	1.5	...	1.00	1.00
Conical	12052.0	1200.0	3.0	3.0	1.5	...	1.00	1.00
Cylinder	16552.0	4500.0	3.0	3.0	1.5	...	1.00	1.00
Conical	18552.0	2000.0	3.0	3.0	1.5	...	1.00	1.00
Cylinder	19352.0	800.0	3.0	3.0	1.5	...	1.00	1.00
Conical	20552.0	1200.0	3.0	3.0	1.5	...	1.00	1.00
Cylinder	24052.0	3500.0	3.0	3.0	1.5	...	1.00	1.00
Conical	26052.0	2000.0	3.0	3.0	1.5	...	1.00	1.00

**External Pressure Calculations:**

From	To	External Actual T. mm.	External Required T. mm.	External Design Pressure bars	External M.A.W.P. bars
10	Ring	3	2.64879	0.1	0.13736
Ring	Ring	3	1.98494	0.1	0.28018
Ring	Ring	3	1.98708	0.1	0.27946
Ring	Ring	3	0.47257	0.1	2.80031
Ring	20	3	0.57319	0.1	2.56958
20	Ring	3	1.86049	0.1	0.32853
Ring	30	3	2.24961	0.1	0.20495
30	Ring	3	No Calc	0.1	No Calc
Ring	40	3	1.62103	0.1	0.46793
40	Ring	3	2.25778	0.1	0.20299
Ring	Ring	3	1.85593	0.1	0.33169
Ring	50	3	0.51366	0.1	2.52828
50	Ring	3	0.56956	0.1	2.4333
Ring	Ring	3	0.50876	0.1	2.55869
Ring	Ring	3	1.80715	0.1	0.35777
Ring	Ring	3	1.80239	0.1	0.36038
Ring	60	3	1.89031	0.1	0.31681
60	Ring	3	0.51049	0.1	2.54788
Ring	Ring	3	2.27117	0.1	0.19962
Ring	70	3	0.6303	0.1	1.95727
70	Ring	3	1.56614	0.1	0.51
Ring	Ring	3	0.62816	0.1	1.96585
Ring	80	3	0.62816	0.1	1.96585
80	Ring	3	0.63409	0.1	1.9422
Ring	Ring	3	1.85593	0.1	0.33169
Ring	90	3	0.51366	0.1	2.52828
90	Ring	3	1.04348	0.1	1.44208

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Ring	Ring	3	1.78809	0.1	0.3684
Ring	100	3	1.96728	0.1	0.28568
100	Ring	3	0.51198	0.1	2.53866
Ring	Ring	3	1.94504	0.1	0.2941
Ring	110	3	0.34821	0.1	4.03247

**External Pressure Calculations:**

From	To	Actual Length Bet. Stiffeners mm.	Allowable Length Bet. Stiffeners mm.	Ring Inertia Required cm**4	Ring Inertia Available cm**4
10	Ring	3958	5366.66	No Calc	No Calc
Ring	Ring	1988	5367.77	4.74696	17.6358
Ring	Ring	1993	5367.78	3.17821	17.6358
Ring	Ring	12.9998	362.283	1.60148	17.6358
Ring	20	100	3105.85	0.090213	17.6358
20	Ring	1679.94	1679.94	No Calc	No Calc
Ring	30	1752.3	1752.3	2.74012	118.645
30	Ring	No Calc	No Calc	No Calc	No Calc
Ring	40	800	3576.41	1.50003	18.5486
40	Ring	1752.3	1752.3	No Calc	No Calc
Ring	Ring	1085.13	1085.13	5.32031	21.7628
Ring	50	0.99994	0.99994	1.08257	17.7314
50	Ring	90	2782.45	No Calc	No Calc
Ring	Ring	13	332.63	0.1026	17.8679
Ring	Ring	1410	4840.94	1.41706	17.8679
Ring	Ring	1400	4840.92	2.80393	17.8679
Ring	60	1587	4841.33	2.96283	17.8679
60	Ring	0.99994	0.99994	No Calc	No Calc
Ring	Ring	1809.08	1809.08	1.80308	17.8679
Ring	70	0.99989	0.99989	3.39301	18.5482
70	Ring	737	3576.31	No Calc	No Calc
Ring	Ring	13	979.513	1.41118	18.5486
Ring	80	50	1799.98	0.11813	18.5486
80	Ring	0.99989	0.99989	No Calc	No Calc
Ring	Ring	1085.13	1085.13	2.0298	18.381
Ring	90	0.99994	0.99994	1.08257	17.7021
90	Ring	375	4840.98	No Calc	No Calc
Ring	Ring	1370	4840.85	1.73585	17.8679
Ring	100	1755	4841.66	3.11292	17.8679
100	Ring	0.99994	0.99994	No Calc	No Calc
Ring	Ring	1679.26	1679.26	1.67377	21.4269
Ring	110	0.99982	0.99982	0.52653	20.0339

**Factored Loads:****Un-Factored Loads:****Weights:**

Fabricated - Bare W/O Removable Internals	3201.3	kg.
Shop Test - Fabricated + Water ( Full )	44267.2	kg.
Shipping - Fab. + Rem. Intls.+ Shipping App.	3201.3	kg.
Erected - Fab. + Rem. Intls.+ Insul. (etc)	3201.3	kg.
Empty - Fab. + Intls. + Details + Wghts.	3201.3	kg.
Operating - Empty + Operating Liquid (No CA)	3201.3	kg.
Field Test - Empty Weight + Water (Full)	44267.2	kg.

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